

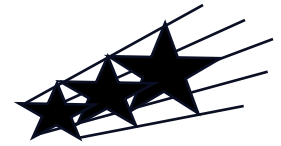
**SYZYGY Engineering**

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# **Key Technologies Regarding Distributed and/or Collaborating Satellite Systems**

**Will Ivancic**  
**SYZYGY Engineering**  
**[ivancic@syzygyengineering.com](mailto:ivancic@syzygyengineering.com)**  
**(440) 503-4892**

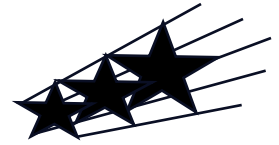
# Outline



**SYZYGY Engineering**

- Network Centric Design
- Policy
- Satellite Communication Basics
- Orbital Mechanics
- Satellite Architectures
- Fractionated Spacecraft
- Cooperative Spacecraft
- Protocols
  - Transport Protocols
  - Store, Carry and Forward

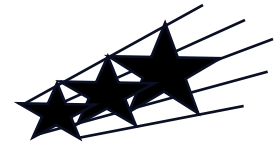
# Note Well



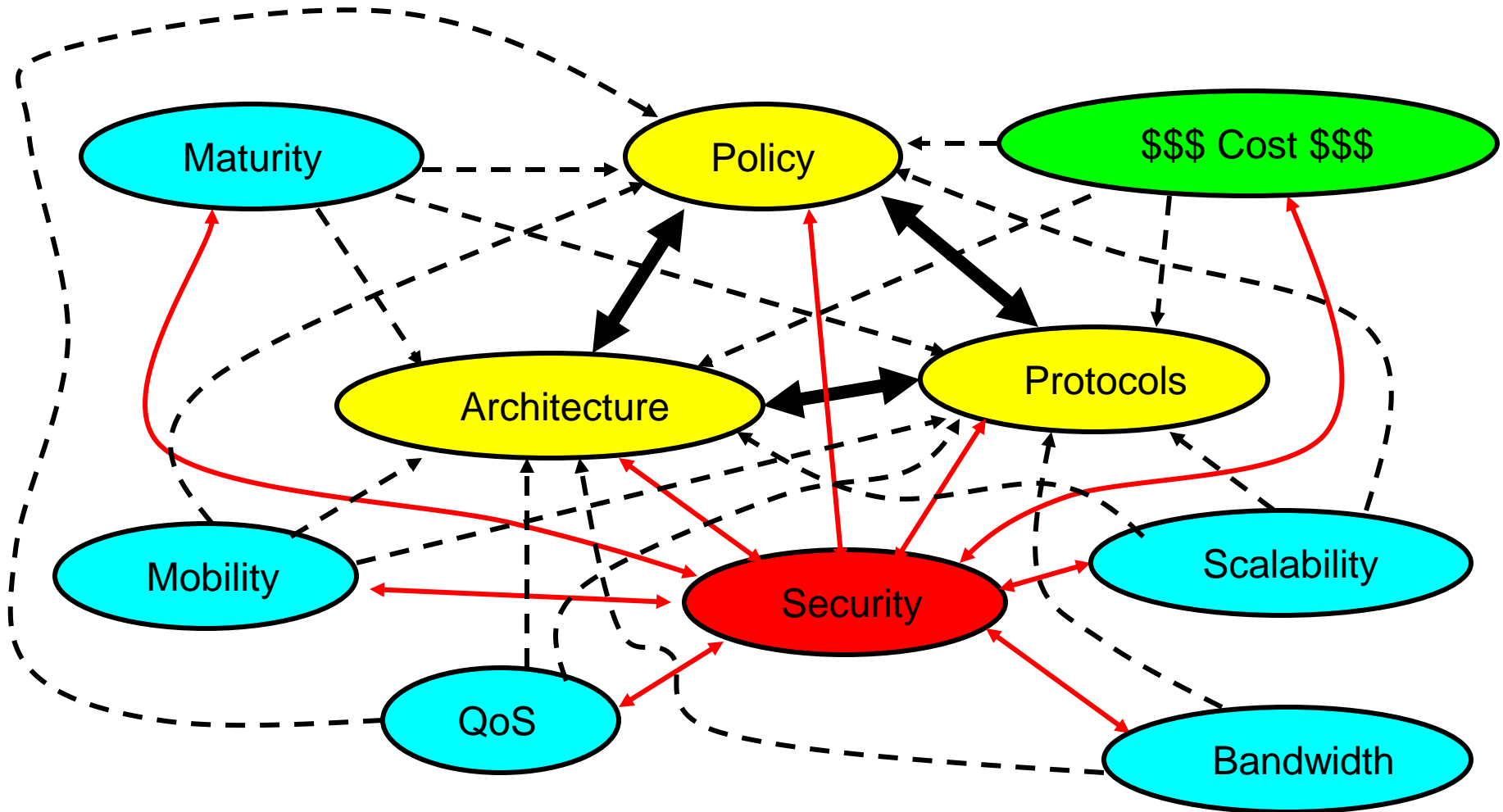
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- There is no free lunch.
- It is hard to develop simple, elegant solutions.
- Not much is completely new – it has probably been thought of before.
  - Technology may not have been capable of implementing the original concepts
- It is always easier to do things on the ground than in space
  - Power, processing, radiation, size, mass, vacuum, heat dissipation.
- Do not believe everything you read – particularly if there is a marketing aspect.

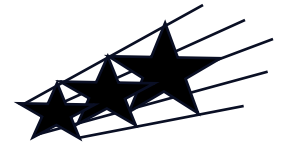
# Network Design Triangle



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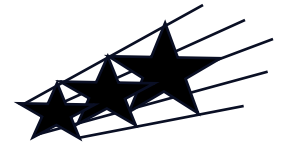
# Policy



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- **Understand Policy (rules of engagement and legal issues).**
  - **It may be the most important thing you do.**
  - **Policy may be the hardest thing to change.**
- **Security has numerous Policy Issues**
- **Spectrum has numerous policy issues**

# Policy Example

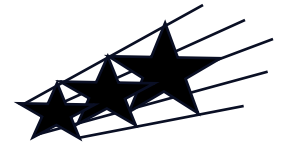


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## SPECTRUM ALLOCATIONS IN MALAYSIA

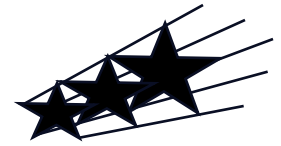


# Spectrum



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- All transmitters have to be registered.
- Receivers do not have to be registered.
- A satellite provider must register the transmitters and should register as to what ground stations it will be transmitting to.
- A ground station should register which satellites it will be communicating with.
  - If the ground stations are in different countries, the transmitters are registered in that country.
- For CubeSats in the amateur band
  - The cubesat can transmit according to the country's regulations that is transmitting over so long as it is within an acceptable power level
  - The ground station uplink is performed under an amateur radio license. Each ground station that transmits is responsible for that transmission under its license.
  - See GENSO - Global Educational Network for Satellite Operations



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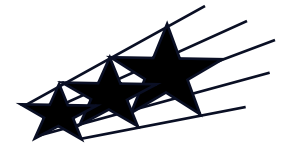
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# **Satellite Communications**

**95% of what you need to  
understand in two slides**



# Electromagnetic Communications

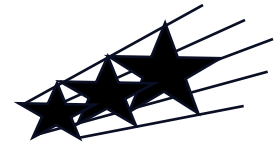


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$$G = \frac{4\pi A_{eff}}{\lambda^2} \quad P_r = \frac{A_t A_r}{r^2 \lambda^2} P_t \quad HPBW = \frac{\lambda}{D}$$
$$c \gg 299\,792\,458 \text{ m/s}$$

- The larger the apertures (area) the higher the gain
- The gain increases as the square of the frequency,  $\nu$  where  $\nu = 1/\lambda$ 
  - This is why Optical is so attractive.
- Larger antenna Diameter (D) and higher frequency results in smaller angular beam widths (equation is for parabolic Antenna, but the generic concept applies)
  - This is why pointing becomes an issue with Optical (lasers)
- Power falls off as the square of the distance from the source and the square of the frequency
- One way trip time to GEO (35,786 km) is approximately 120 msec with a Round Trip Time of approximately 500 msec
- One way trip time to LEO (400 km) is 1.3 msec with a RTT of approximately 5.3 msec

# Electromagnetic Communications

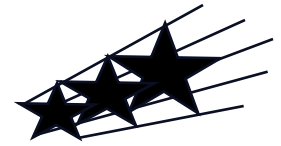


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$$C = B \log_2 \left( 1 + \frac{S}{N} \right) = B \log_2 \left( 1 + \frac{E_b}{N_0} \frac{B}{f_b} \right) \cdot \frac{S}{P_r}$$

$$P_r(\text{dbm}) = P_t(\text{dbm}) + \text{Gains}(\text{dbm}) - \text{Losses}(\text{dbm})$$

- Shannon–Hartley theorem:  $C$  is the maximum rate at which information can be transmitted over a communications channel of a specified bandwidth,  $B$ , in the presence of noise,  $N$ .
  - If I have more Bandwidth,  $B$ , I can transmit more information.
- The higher the energy per bit relative to noise,  $E_b/N_0$ , the lower the bit-error-rate BER.
  - Reducing the transmission data rate,  $f_b$ , improves (reduces) the BER
- Gains include antenna and amplifier gains



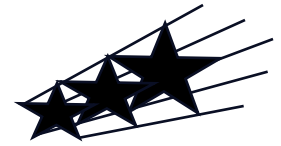
**SYZYGY Engineering**

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# **Orbital Mechanics and Rocket Science**

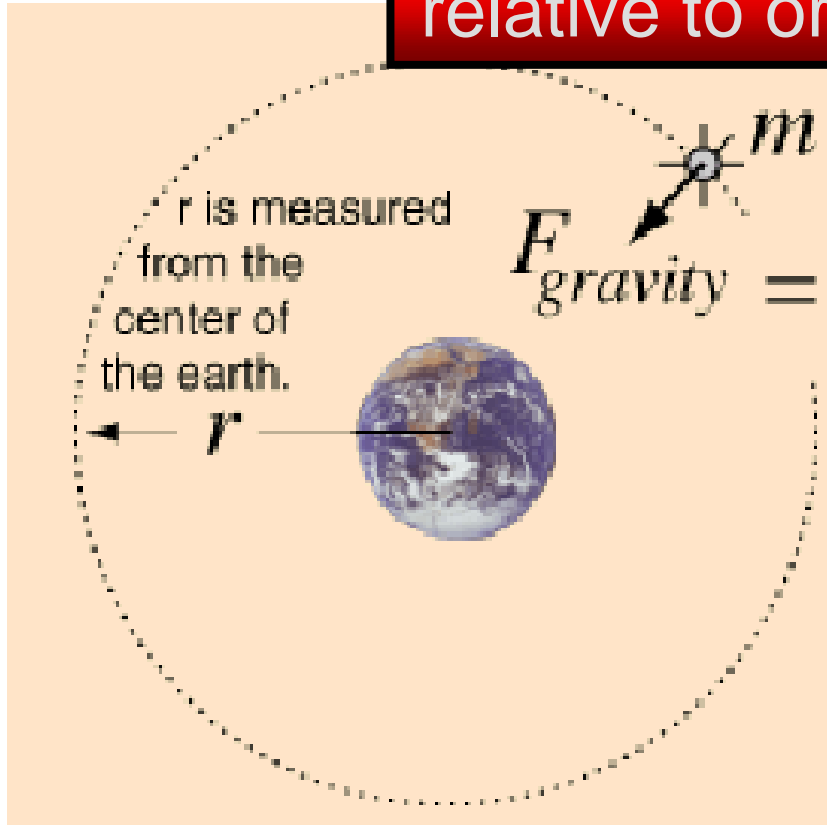
**The keys to understanding  
space communication  
architectures**

# Circular Orbital Equation



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Space is very predictable  
relative to orbital dynamics.

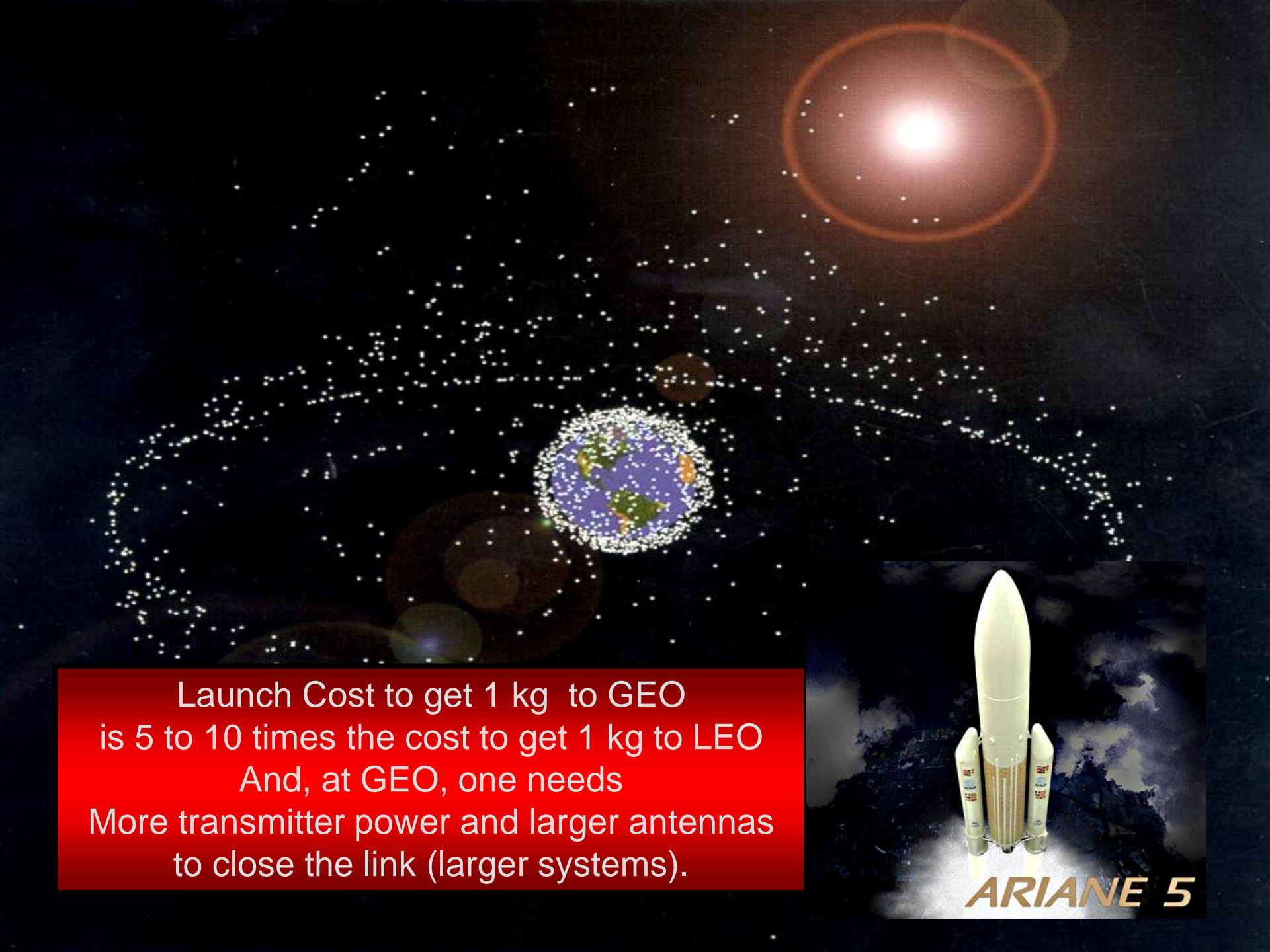


$$v = \sqrt{\frac{GM}{r}}$$

International Space Station travels at  
27,744 kilometers per hour (over 6  
times faster than the fastest bullet)  
resulting in an orbital period of ~ 90  
minutes.

Introduction to Orbits

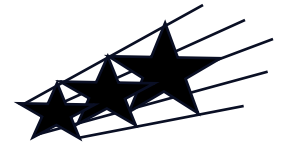
[https://www.youtube.com/watch?v=lekLGgj\\_Wkk](https://www.youtube.com/watch?v=lekLGgj_Wkk)

The background of the slide is a composite image. It features a view of Earth from space, surrounded by a dense cloud of small white dots representing a satellite constellation. A large, bright sun with a prominent orange and red lens flare is positioned in the upper right corner. The overall scene is set against the blackness of space.

Launch Cost to get 1 kg to GEO  
is 5 to 10 times the cost to get 1 kg to LEO  
And, at GEO, one needs  
More transmitter power and larger antennas  
to close the link (larger systems).



**ARIANE 5**



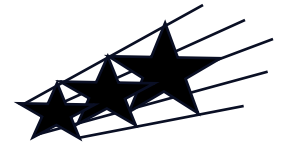
**SYZYGY Engineering**

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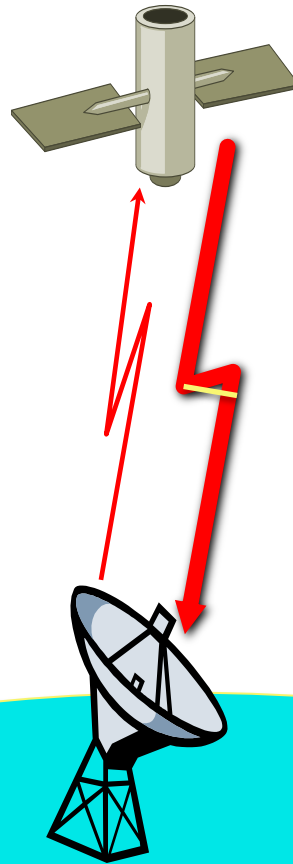
# **Basic Satellite Architectures**

# Often Highly Asymmetric Links

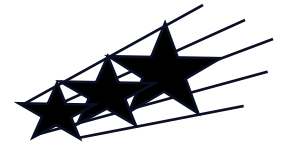
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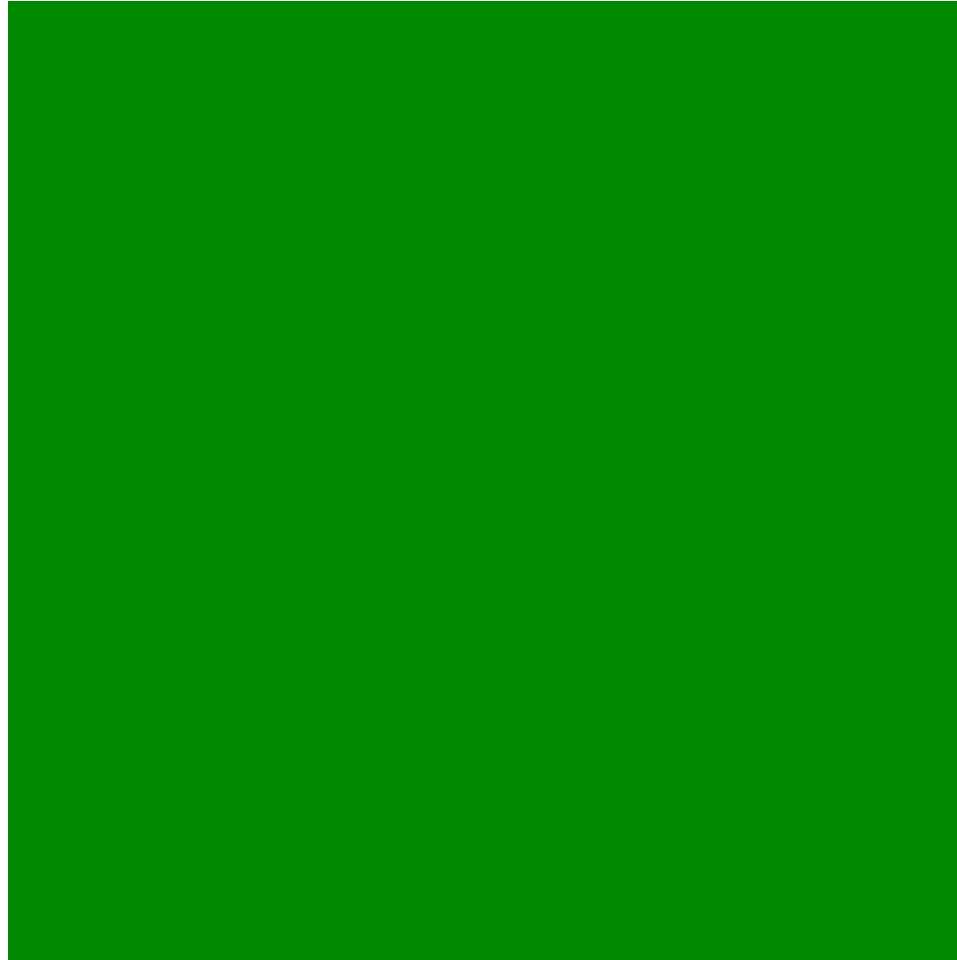
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# Polar Orbit



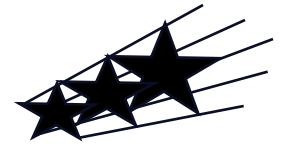
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Source: <http://commons.wikimedia.org/> via youtube

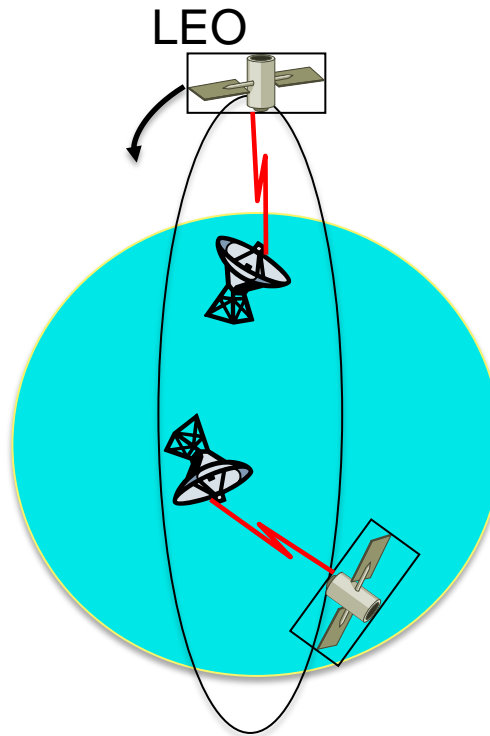


# Low Earth Orbit (LEO) Only

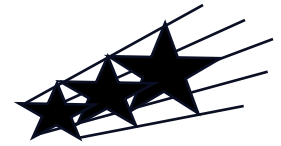


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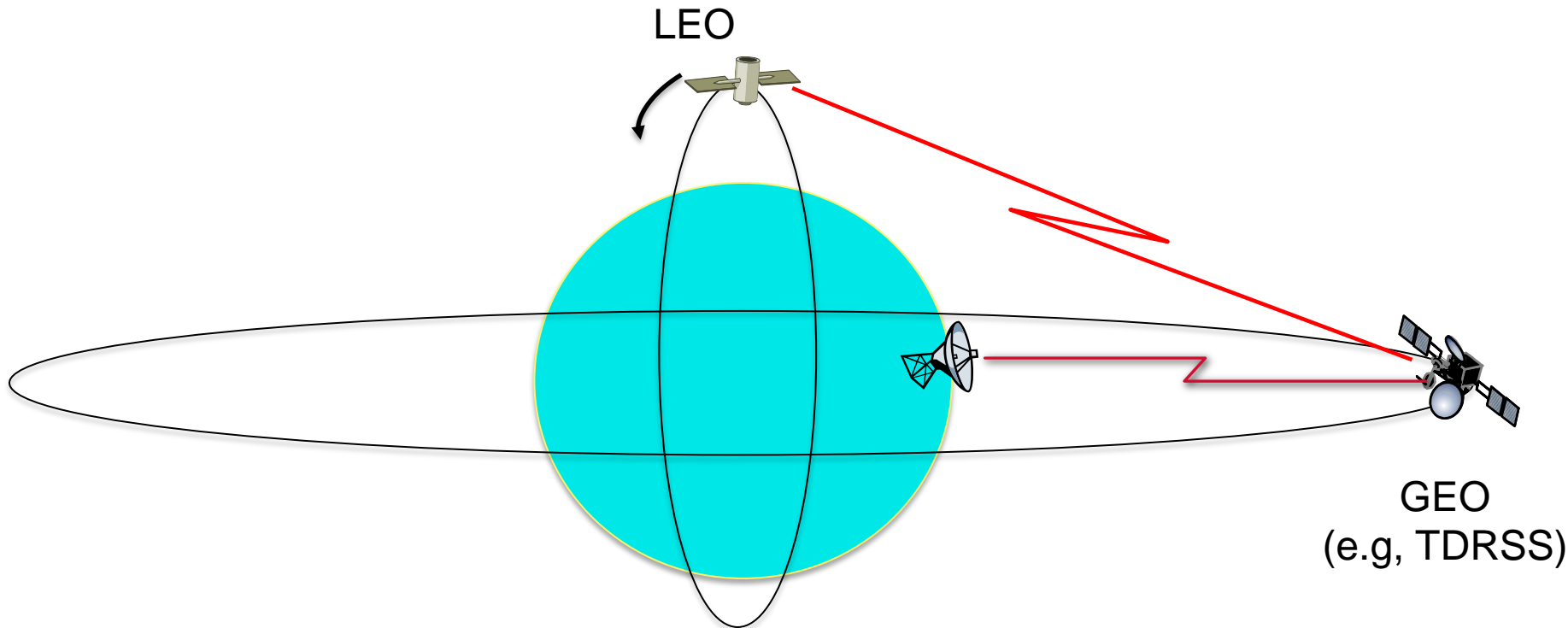
- LEO orbits are below the Earth's **radiation belts**, which allows for use low-cost Commercial-Off-The-Shelf (COTS) components.
- Radiation still must be considered – depending on the duration of flight.



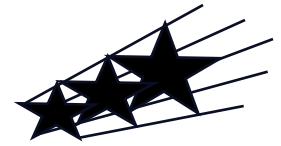
# LEO via Geostationary Orbit (GEO)



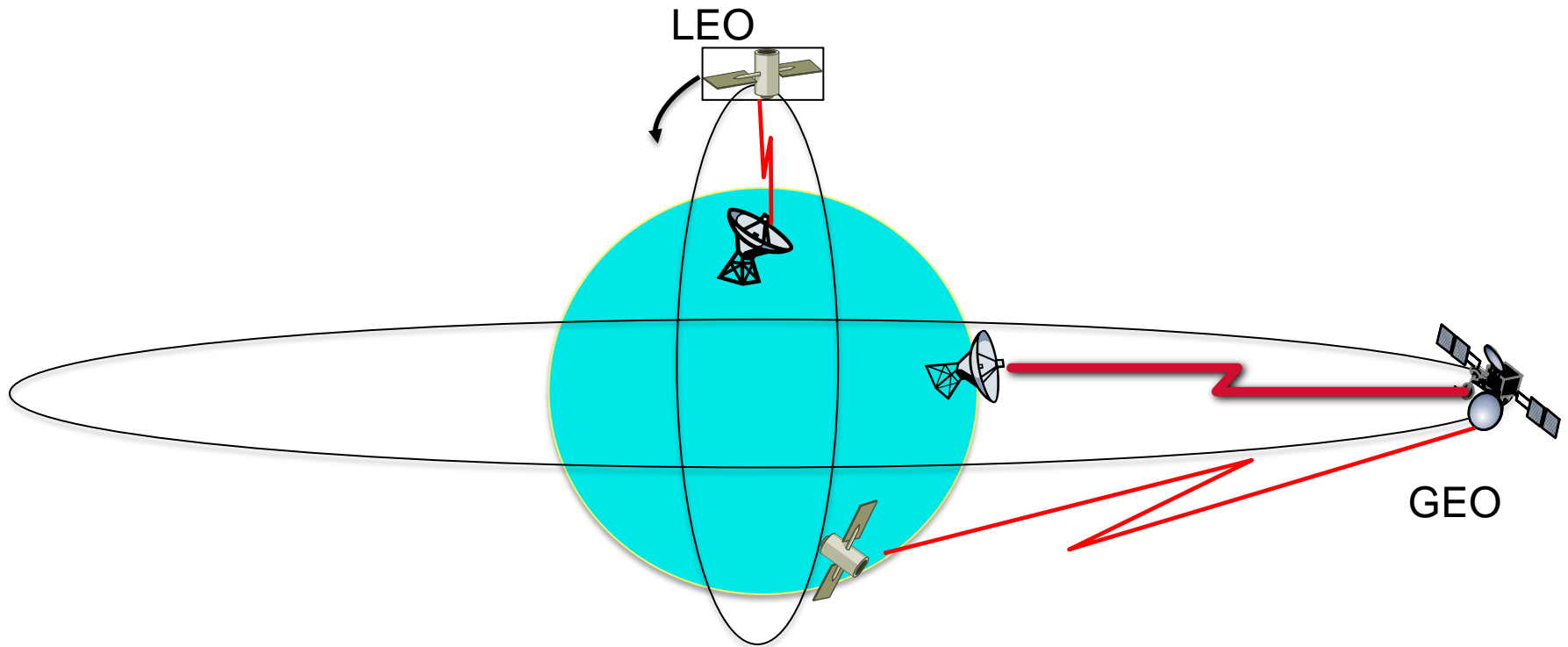
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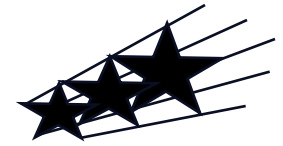
# LEO and GEO



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# Teledesic – 840 Communication Satellites in original plan



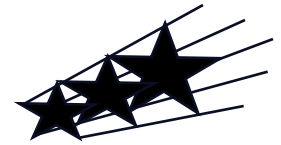
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SaVi

Talk of Google putting up  
1600 Satellites  
Imaging or  
???Communications???

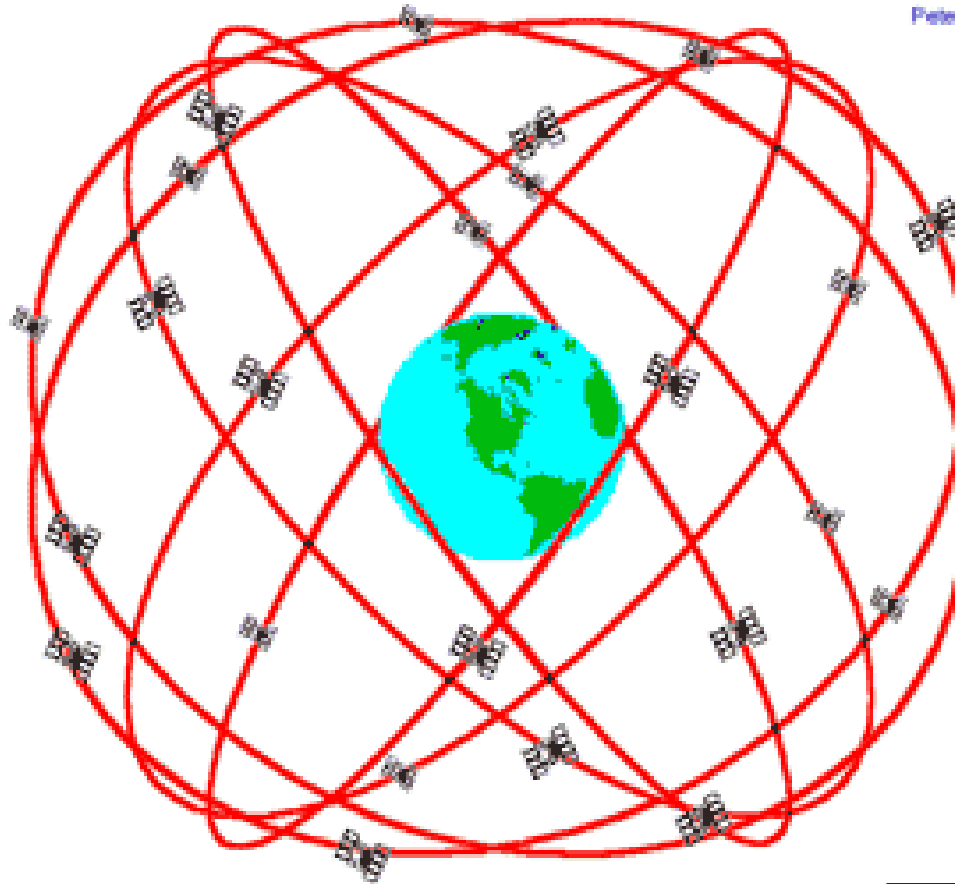
The smaller the beams, the greater the  
frequency reuse. But, these are moving, so  
there are routing and hand-off  
complexities.

# Global Positioning System



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Peter H. Dana 9/23/98



**GPS Nominal Constellation**

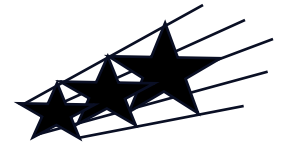
**24 Satellites in 6 Orbital Planes**

**4 Satellites in each Plane**

**20,200 km Altitudes, 55 Degree Inclination**

GPS is in a much  
higher orbit than LEO.

# Stationary LEO!



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**First Facebook and now Google say it will  
use high-flying drones to power the Internet**



**TITAN**  
AEROSPACE





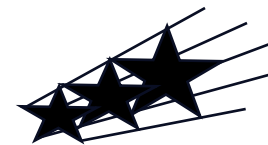
## DARPA F6

Future, Fast, Flexible, Fractionated  
Free-flying Spacecraft United by  
Information Exchange, or System F6  
was Cancelled in 2013

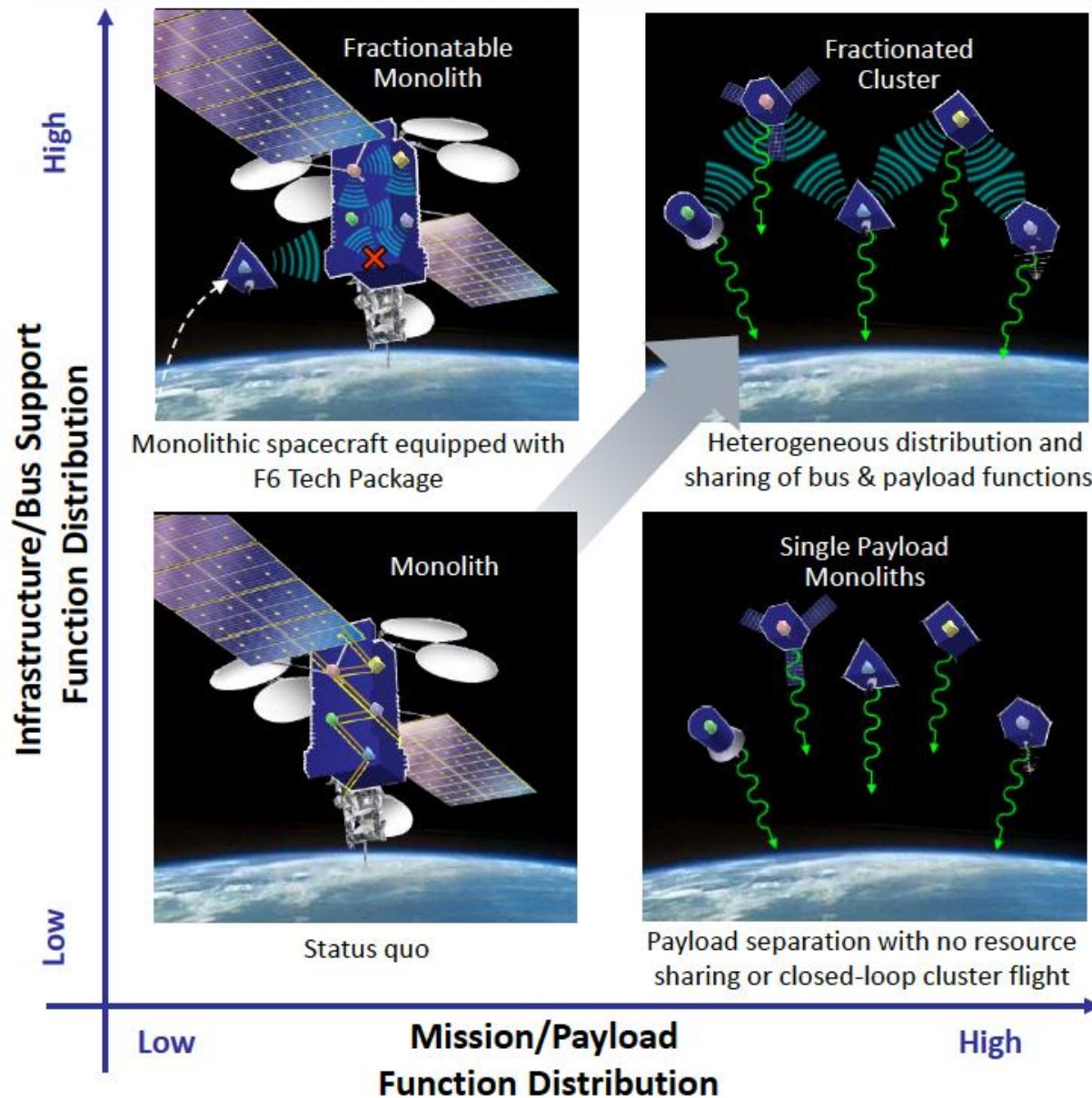
# Fractionated Spacecraft



# Fractionated Space Architectures



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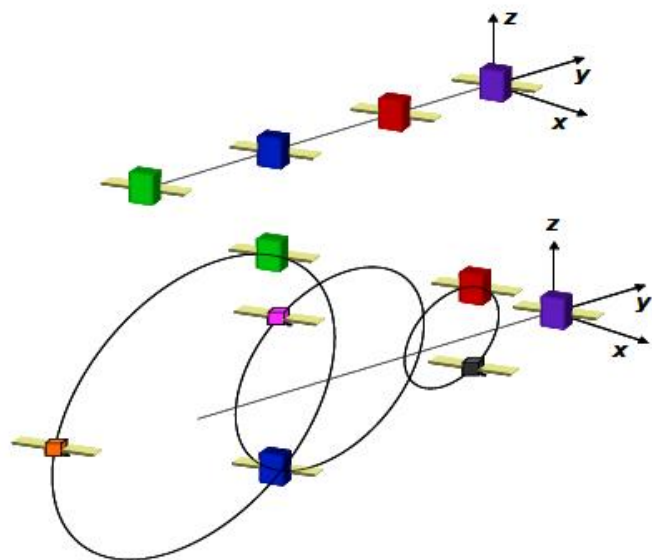
## Enables

- Resource Sharing
  - Power
  - Telecommunication
  - Processing
- Redundancy
- Rapid cluster maneuvering
- Multiple Vendors
- Multi-level Security
- Security Isolation
- Stabilization Isolation (i.e., telescope)
- Build as you go
  - Everything does not have to be ready at the same time

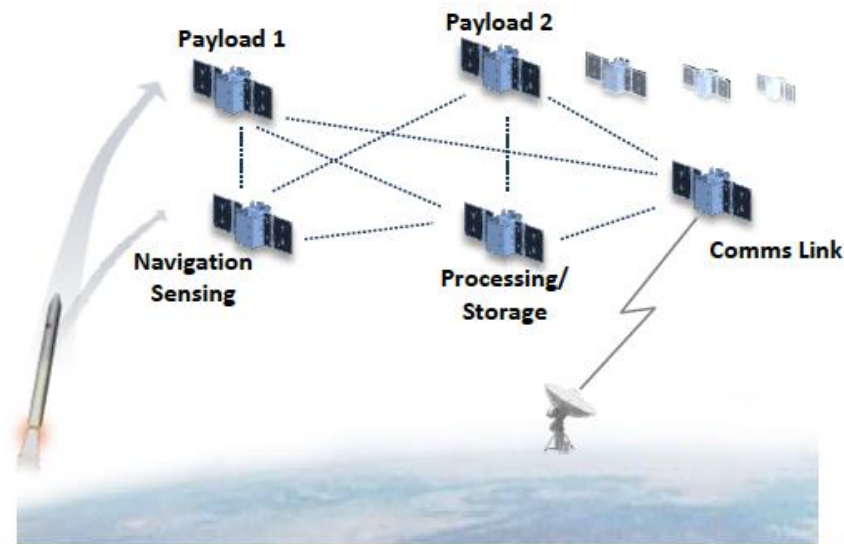




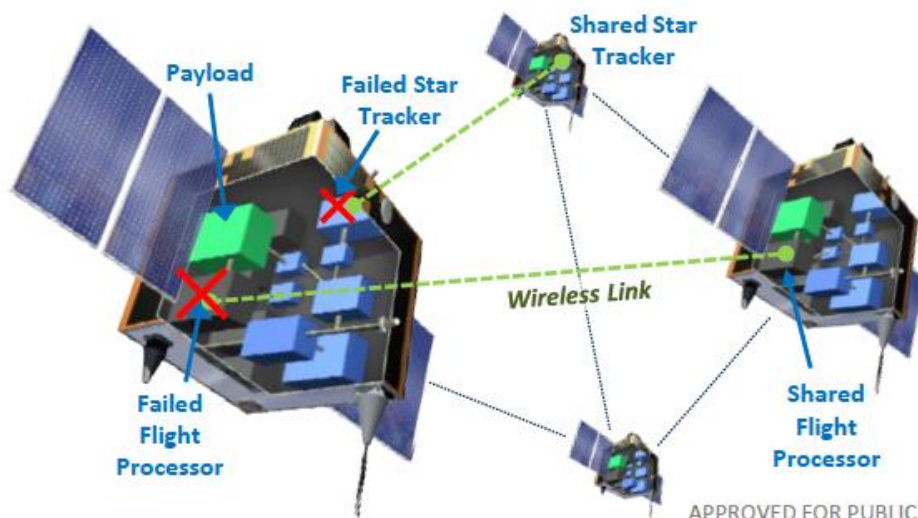
# Key Capabilities for 2015 On-Orbit Demonstration



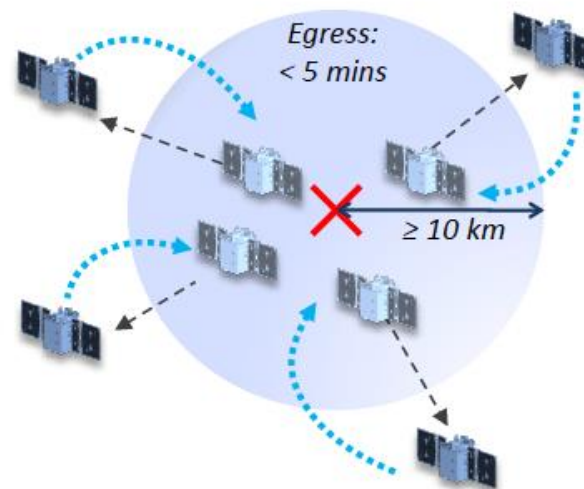
## Demo 2: Resource Sharing at Multiple Security Levels



## Demo 3: Cluster-Level Fault Tolerance

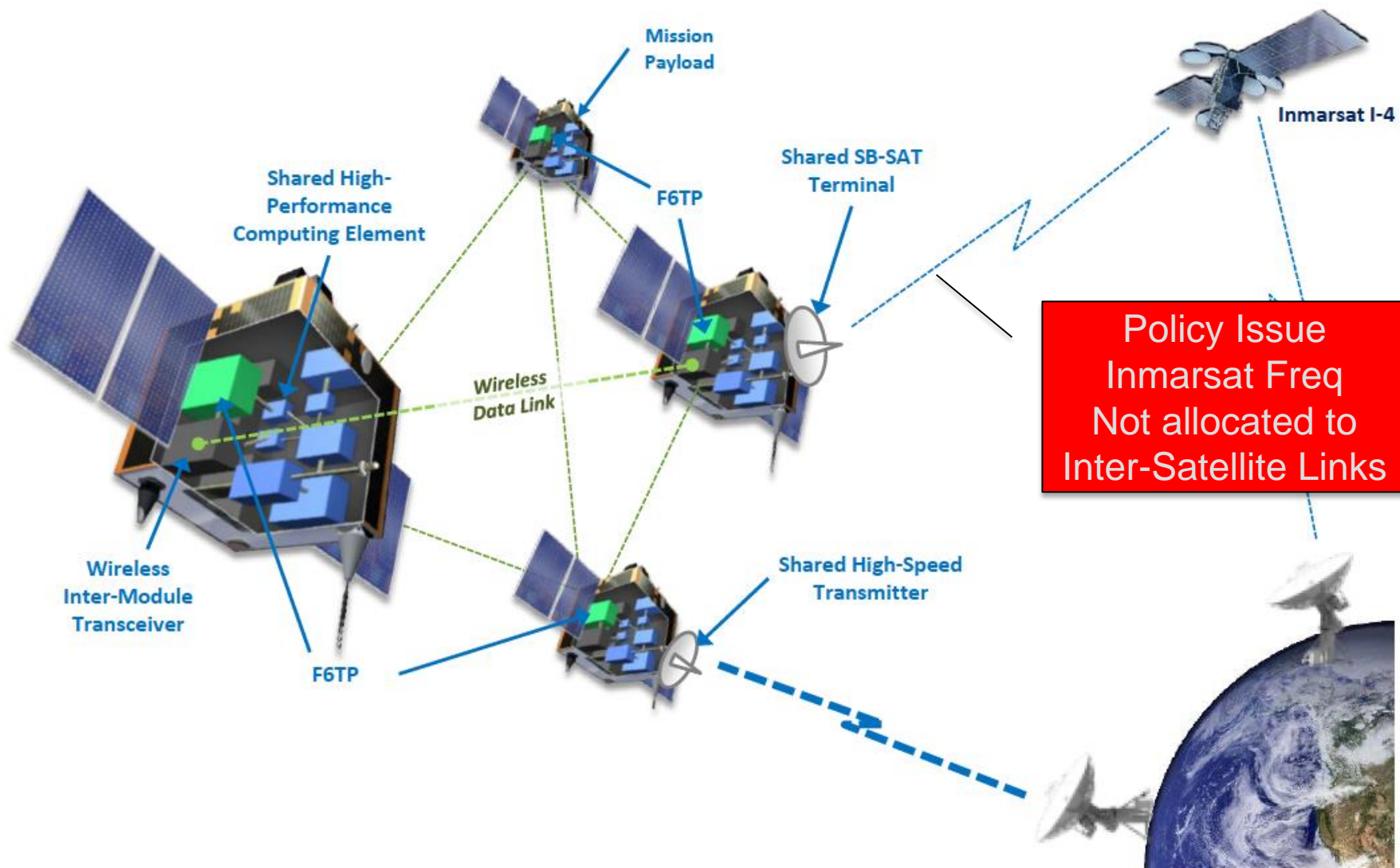


## Demo 4: Defensive Scatter and Re-Gather



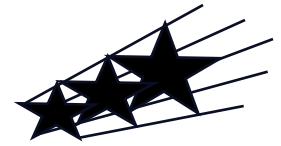


# Notional System F6 On-Orbit Demo



# Space Interferometer

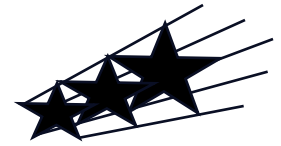
## (Conceptual – Fractionated Instrument)



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- Higher angular resolution in astronomical images requires increasing apertures of telescopes or increasing baselines of interferometers.
- For monolithic units, the mass of the support structure, and propellant for launch is about to exceed technical and financial boundaries.
- Solution to overcome the mass constraint is to combine satellites in autonomous formation flight to behave just like a rigid body.
- Requires precise navigation, control and timing (to correlate data)



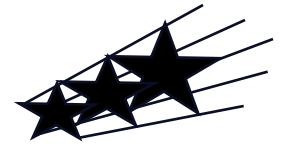
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# **Cooperative Spacecraft**

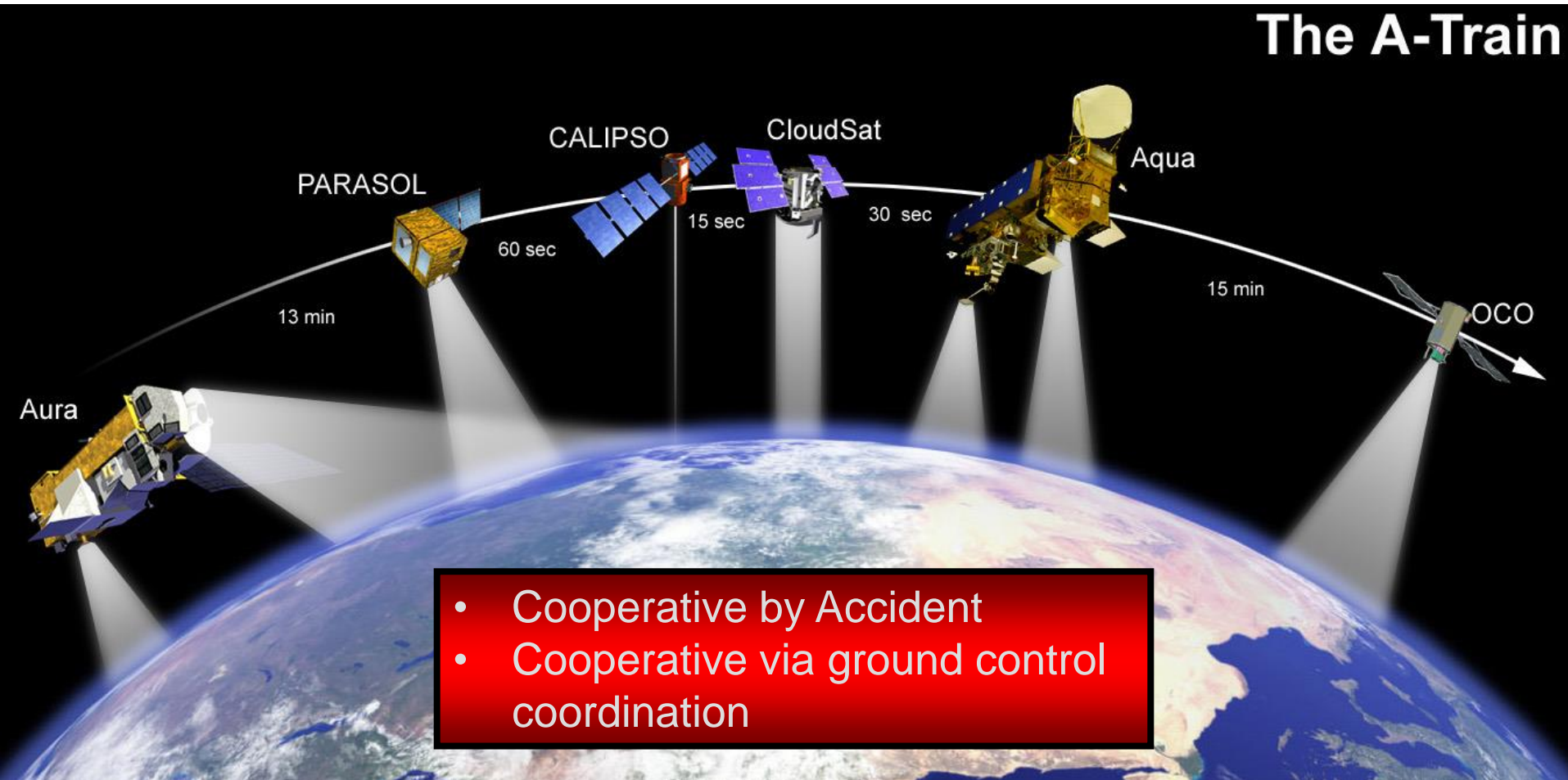


# NASA's A-Train



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## The A-Train



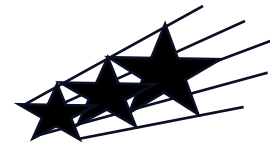
# QB50 Project



Artist rendition of the QB50 CubeSats (courtesy of Ruedeger Reinhard)

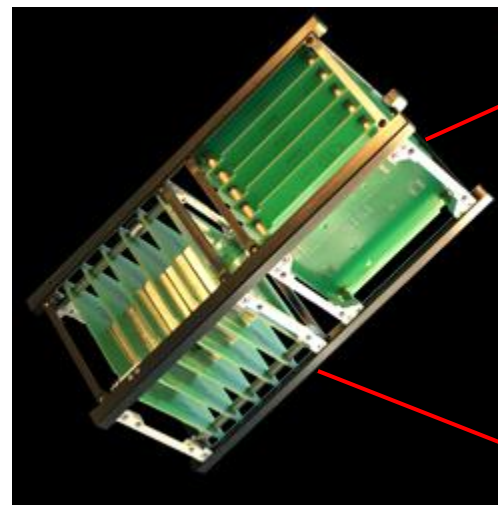


# QB50



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- An international network of 50 CubeSats sequentially deployed – String of Pearls
- Science: Multi-point in-situ long-duration measurements and in-orbit demonstrations in the lower thermosphere
- Initial altitude: 350 km (circular orbit, high inclination)
- Downlink using the QB50 Network of Ground Stations
- On QB50, the CubeSats are the primary payload. On all other missions CubeSats have been secondary payloads.
- Launch scheduled for January 2016



**Science Unit**



**Functional Unit**

Power

Telecommunications

Tech Demonstrations

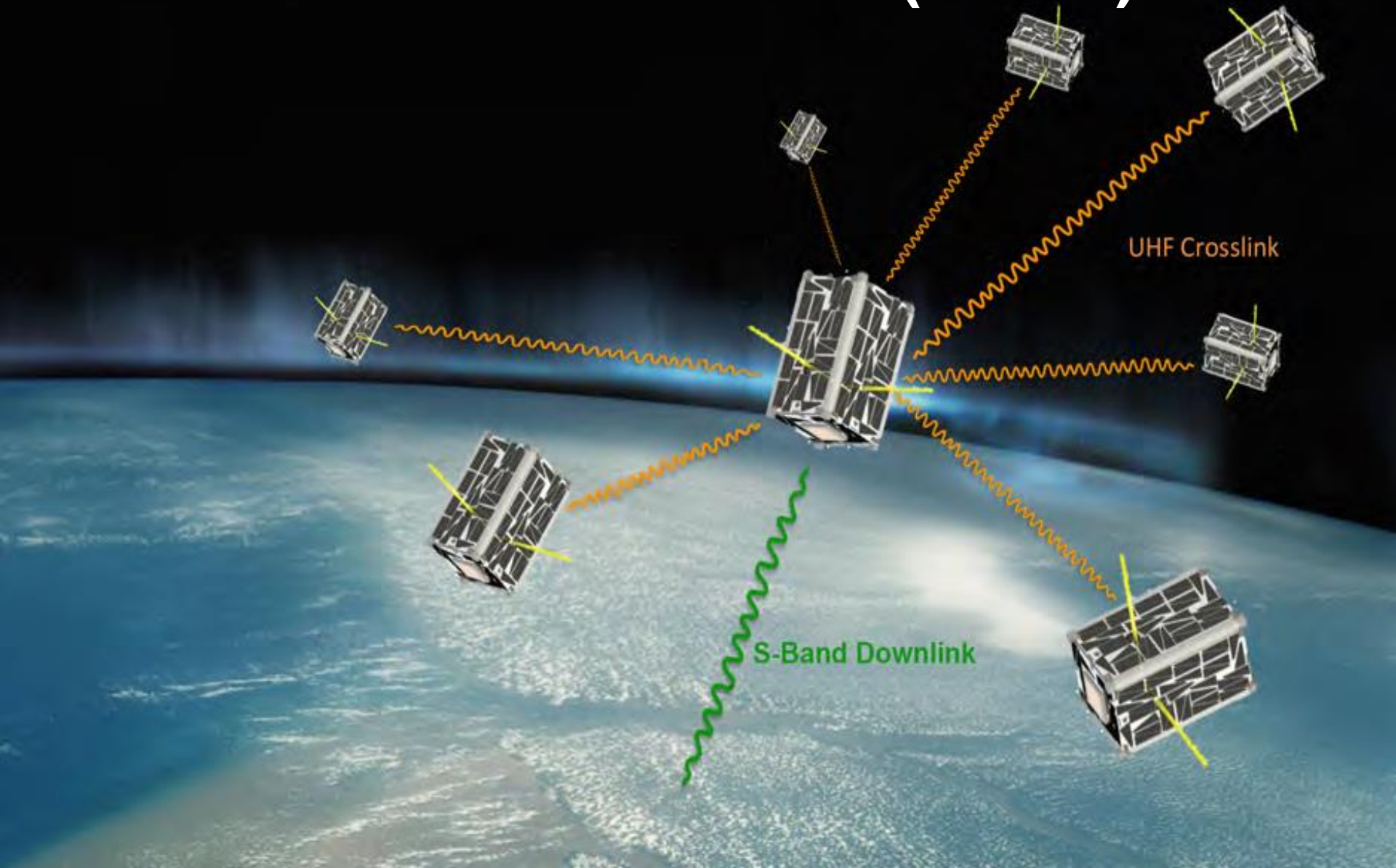
[www.QB50.eu](http://www.QB50.eu)



*von Karman Institute  
for Fluid Dynamics*



# NASA's Edison Demonstration of Smallsat Networks (EDSN)

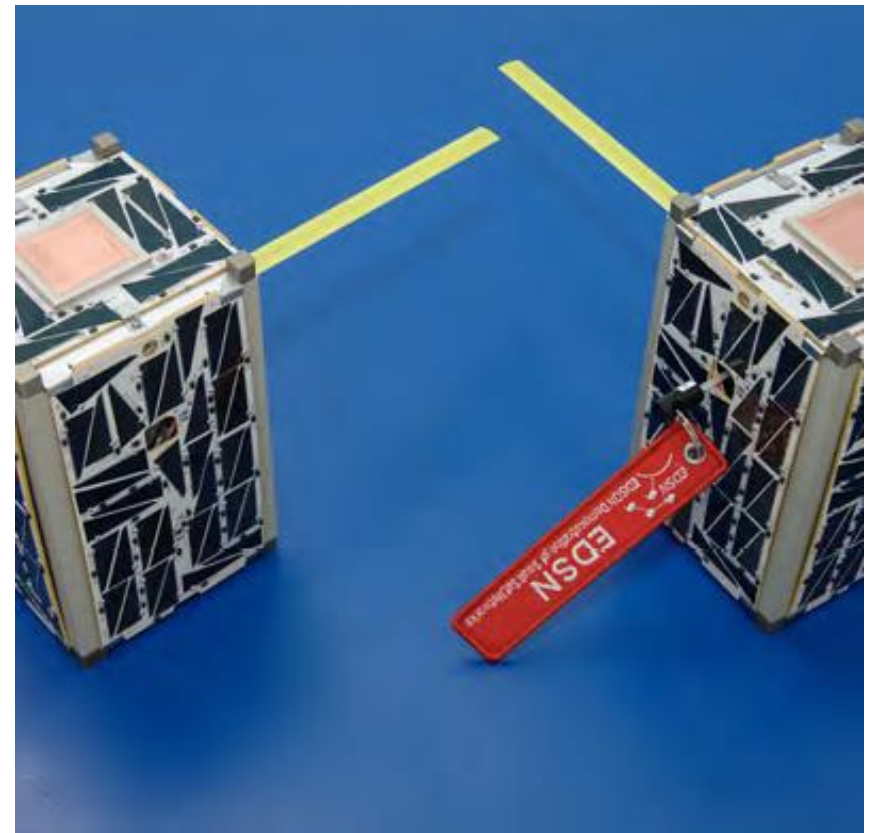




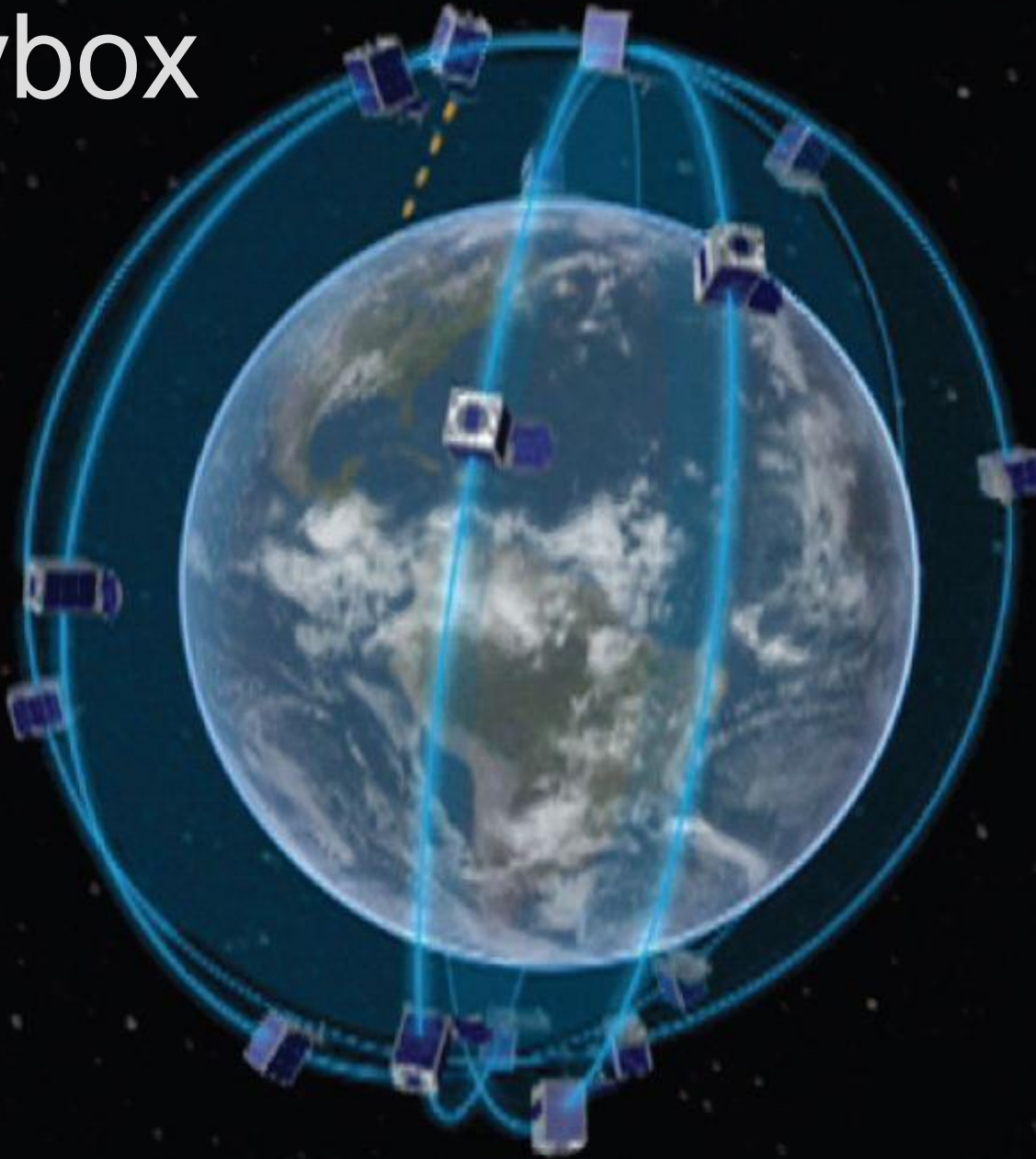
## Enable Cross-Link Communication and Multipoint Physics

- **8 cubesats into a loose formation**
- **Approximately 500 km above Earth**
- **1.5 cubesat unit**
  - about the size of a tissue box
  - weighs approximately 4 pounds (2 kg)
- **60-day operational period in orbit.**
- **Science is to make distributed, multipoint space radiation measurements.**
- **Demonstrate advanced communications, including a network that allows for data to be sent between satellites as needed.**
- **Launch date, end or 2014**

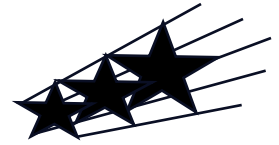
## Completed EDSN Spacecraft



# Skybox



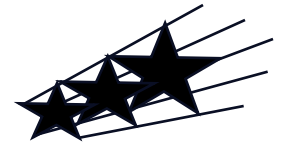
# Skybox



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- **24-satellite constellation at a 500-kilometer orbit**
- **Skybox treats each satellite as just another server in a server farm.**
- **4 different orbital planes**
- **Downlink Feed X-band, 8025-8400 MHz**
- **Uplink Feed S-band, 2025-2110 MHz**
- **Satellite revisit rates of five to 10 times per day**
  - Critical to applications such as change detection
- **High-resolution imagery (1 meter or better) and full-motion video for commercial sale.**
- **Business case is data-mining and analytics**

# Skybox

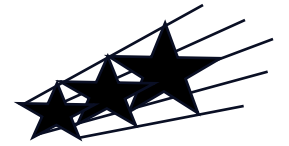


**SYZYGY Engineering**

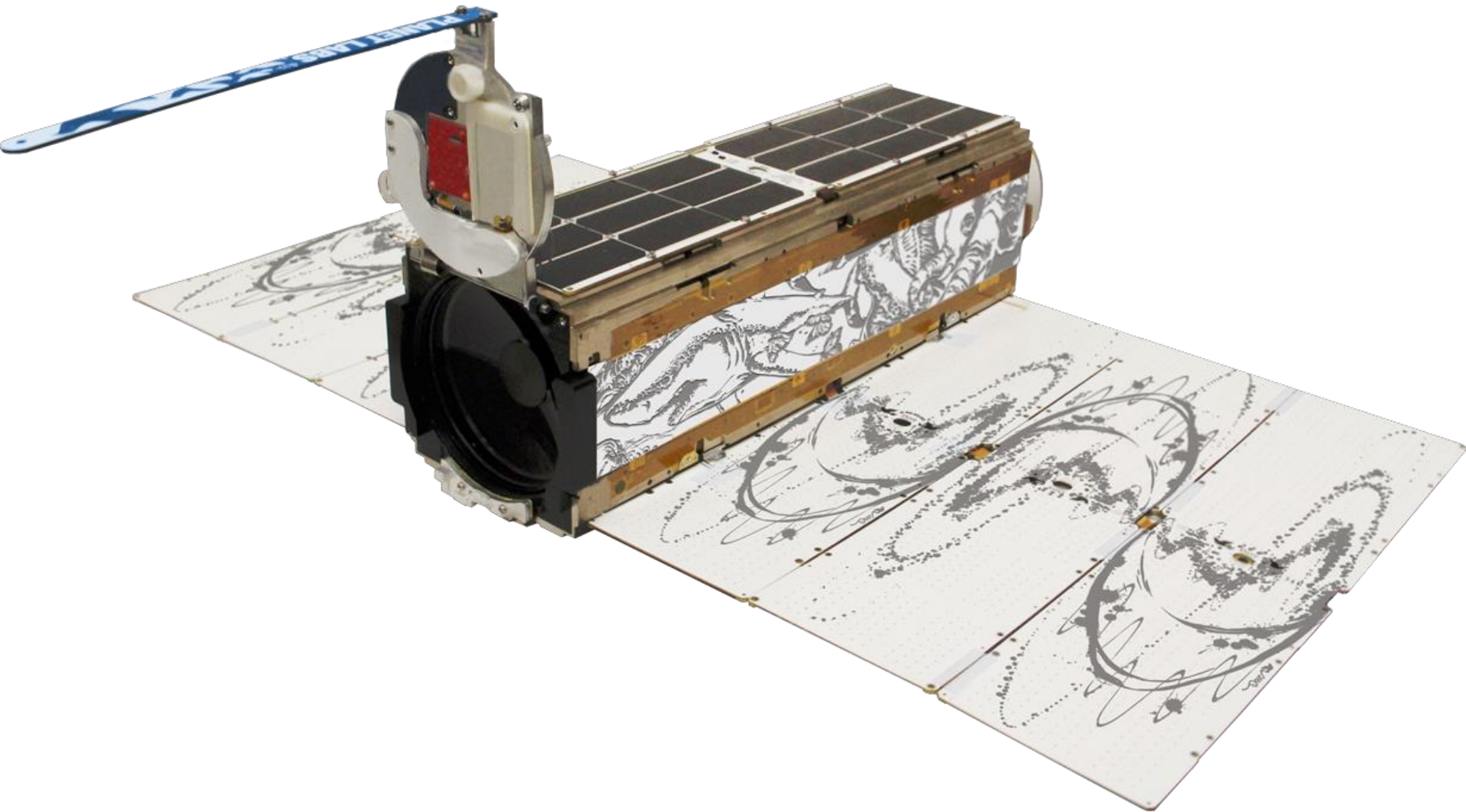




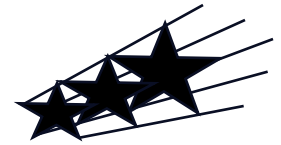
# Planet Lab “Dove” CubeSat



**SYZYGY Engineering**



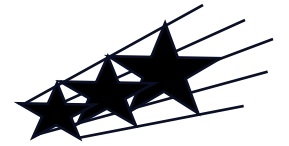
# Planet Lab



SYZYGY Engineering

- Earth-imaging satellite constellation
- 28 Doves currently
- **100 planned**
- Orbit at 400 miles (640 km) and provide imagery with 3-5 m resolution
- Dove 2 Communications
  - UHF
  - MHX-2400 (2.4-2.5 GHz ISM band) Spread Spectrum used 21 meter dish (poor link quality)
  - **X-Band Radio DVB-S2 format used 6 meter dish (good link)**
- Currently launches from the International Space Station
- Fresh data from any place on Earth
- **Business Case: Global sensing and analytics**

# 6 Meter Dish

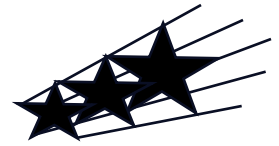


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# RapidEye

(by BlackBridge merged GeoEye and DigitalGlobe ) SYZGY Engineering



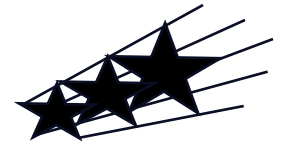
- The largest sub-meter constellation of satellites
  - IKONOS can collect panchromatic and multispectral imagery which can be merged to create 0.82-meter color pan-sharpened imagery.
  - GeoEye-1 acquires 50 cm TRUE\* panchromatic and 2.0 meter multispectral imagery.
  - QuickBird is a 60 cm, 4-band color satellite, and is capable of collecting multispectral and panchromatic imagery.
  - WorldView-1 is a 50 cm TRUE\*, panchromatic satellite and is also capable of collecting in-track stereo imagery.
  - WorldView-2 is a 50 cm TRUE\*, 8-band color satellite capable of collecting 2.0 meter multispectral, panchromatic, and in-track stereo imagery.



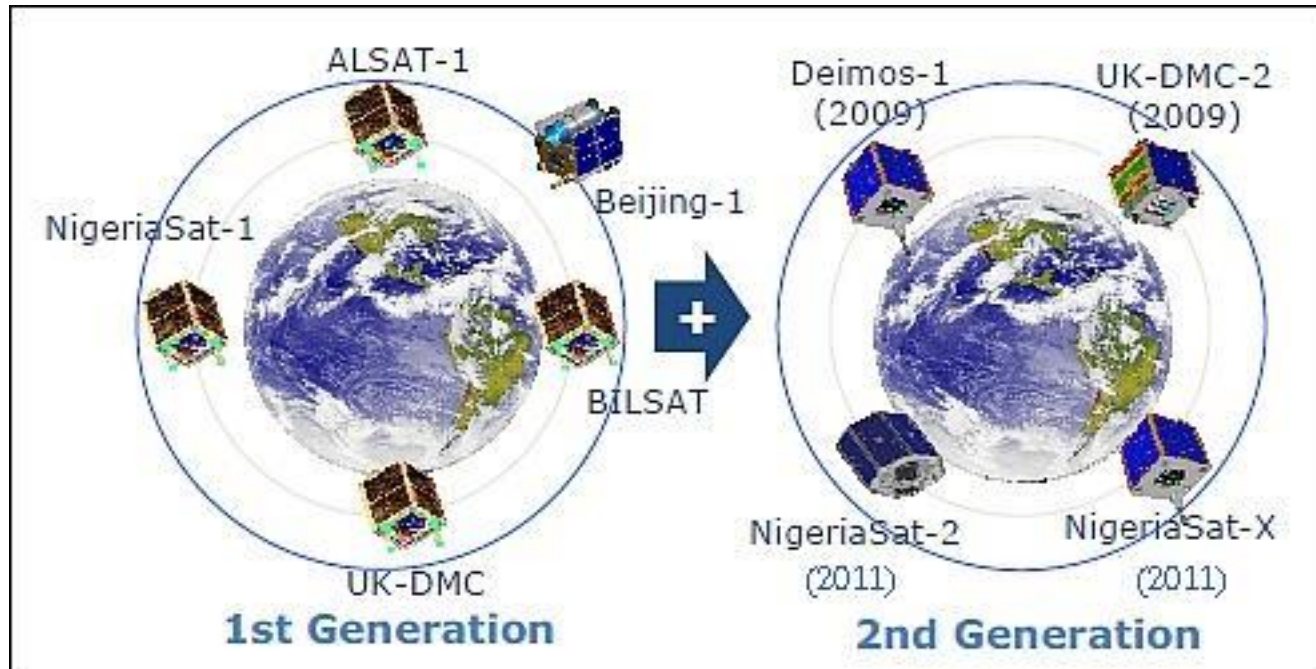


# Disaster Monitoring Constellation

Surrey Satellite Technology LTD



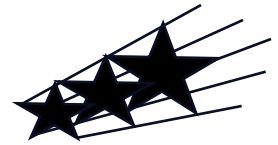
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The sun-synchronous orbits of these satellites are coordinated so that the satellites follow each other around an orbital plane, ascending north over the Equator at 10:15 am local time (and 10:30 am local time for Beijing-1).

# Virtual Mission Operations Center

[http://roland.grc.nasa.gov/~ivancic/papers\\_presentations/papers.html](http://roland.grc.nasa.gov/~ivancic/papers_presentations/papers.html) **SYZYGY Engineering**



Screenshot of a web application interface for a Virtual Mission Operations Center, displayed in a Mozilla Firefox browser window.

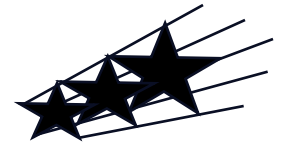
The browser window shows the address bar with the URL `http://67.135.22...pplication.html`. The page title is "UNCLASSIFIED". The user is logged in as "Steven Sage" and can sign out. The interface includes a map of a region in Mississippi, showing roads, rivers, and towns. A green rectangle highlights a specific area on the map, and a blue rectangle highlights another area. The map is labeled with "Seminary", "Moselle", "Ovett", "Sumrall", "Petal", "Hattiesburg", and "Cole".

An "Analysis Results" window is open, displaying a table of mission data:

Mission	Start Time	Duration
GeoEye-1	13204840Z Feb13	4m 6s
GeoEye-1	13222521Z Feb13	5m 3s
GeoEye-1	14092133Z Feb13	6m 59s
GeoEye-1	14212234Z Feb13	6m 58s
GeoEye-1	15082015Z Feb13	5m 52s
GeoEye-1	15095843Z Feb13	3m 22s
IKONOS-2	13213627Z Feb13	7m 15s
IKONOS-2	14083356Z Feb13	6m 27s
IKONOS-2	14221158Z Feb13	6m 31s
IKONOS-2	15090839Z Feb13	7m 17s
QuickBird2	13211152Z Feb13	1m 8s

The "Analysis Results" window includes a "Close" button.

# Protocols



SYZYGY Engineering

Protocols are tools for communication



Some should not be used.



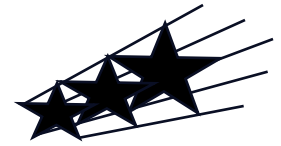
Some work OK



Some are designed for the job at hand

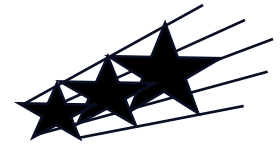
**One size does not fit all.**





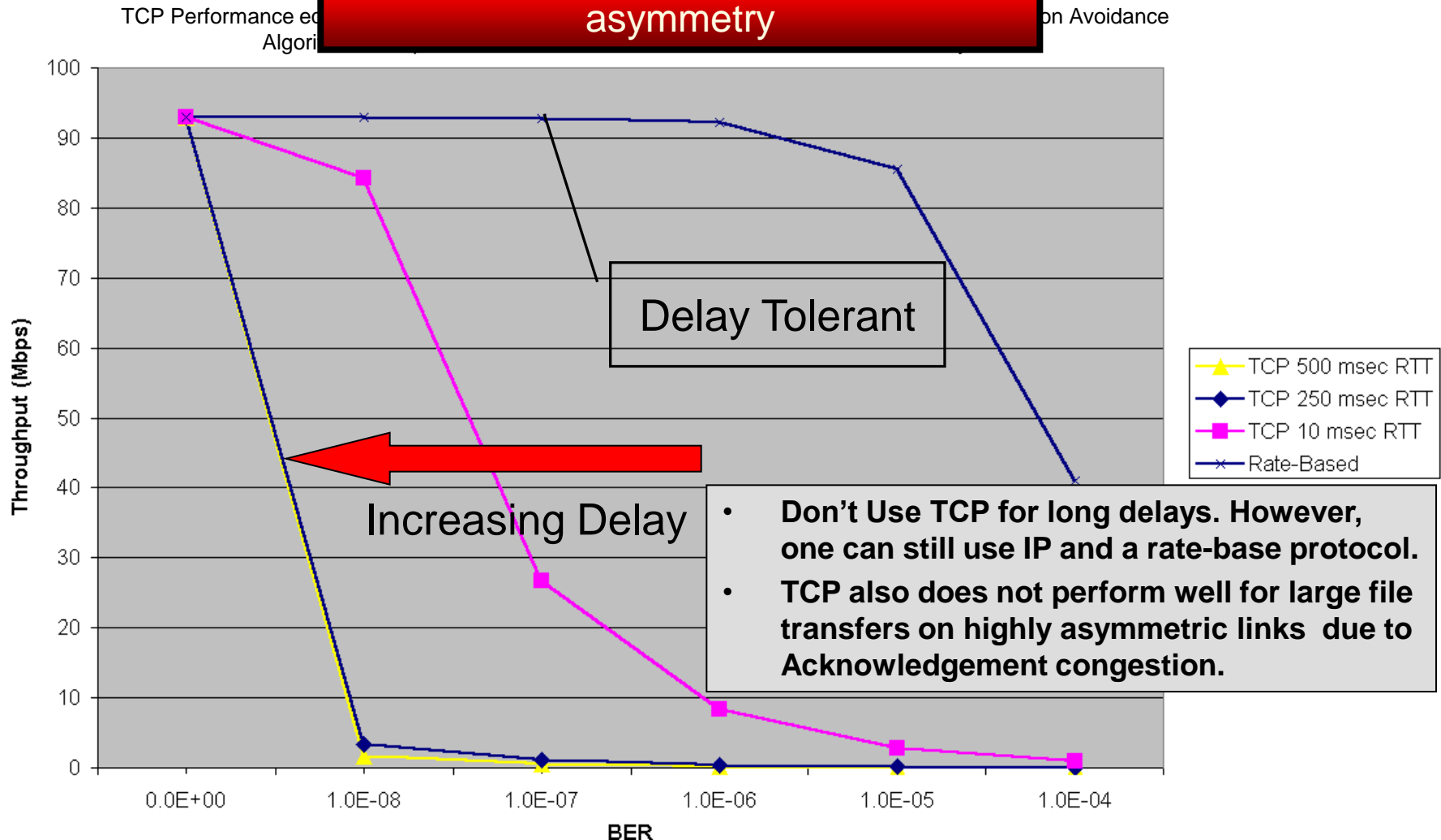
# **Reliable, High-Rate Transport Protocols for Highly Asymmetric Private Links**

# Theoretical Steady State Throughput

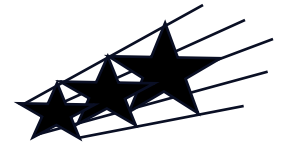


SYZYGY Engineering

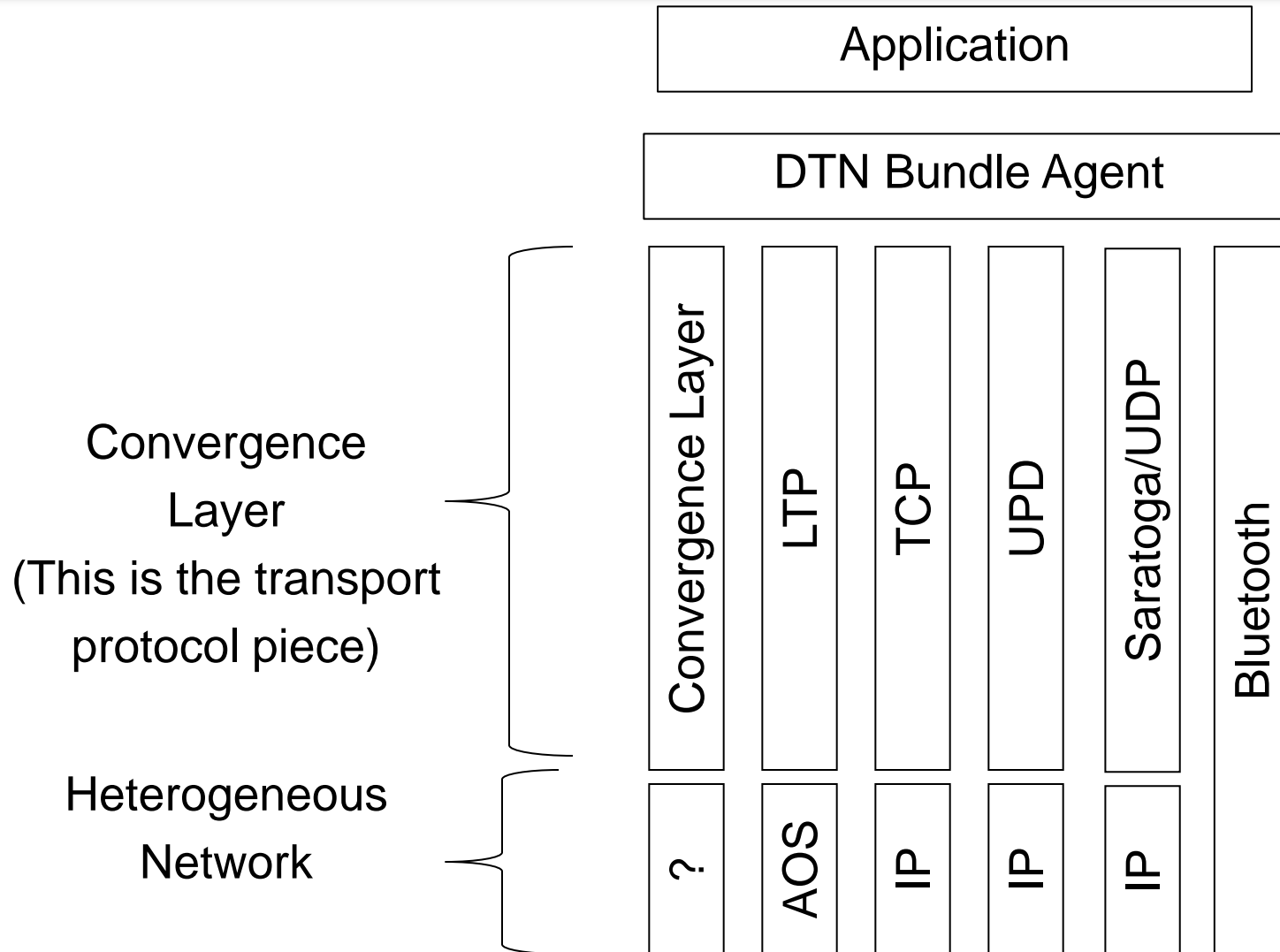
LEO satellites do not have long  
propagation delay, but do have high  
asymmetry



# DTN Layering (RFC 5050)

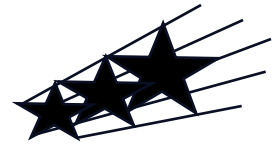


SYZYGY Engineering



# **“PROTOCOLS FOR STORE-AND-FORWARD MESSAGE SWITCHING VIA MICROSATELLITES”**

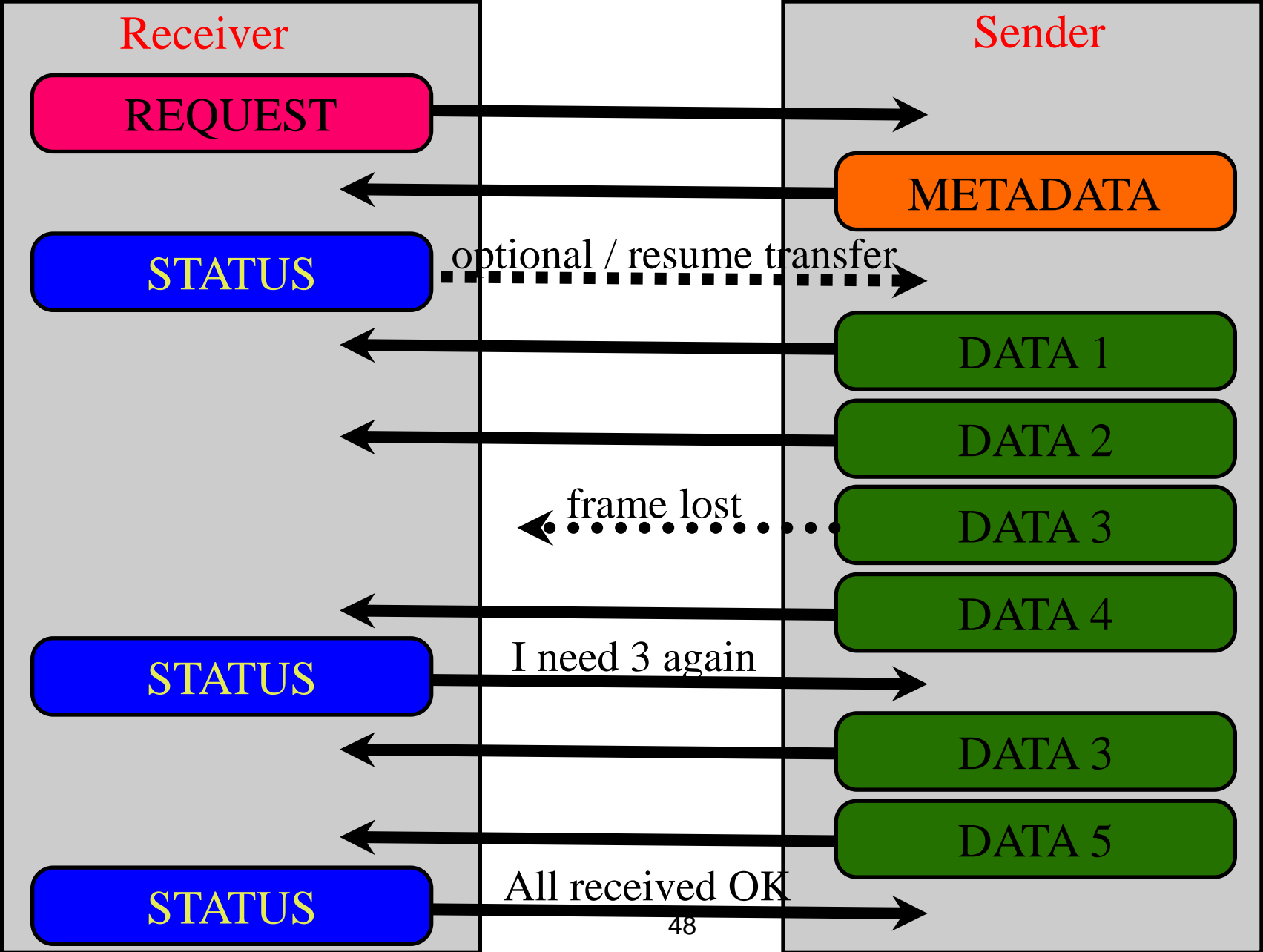
J. W. Ward and H. E. Price - 1991



**SYZYGY Engineering**

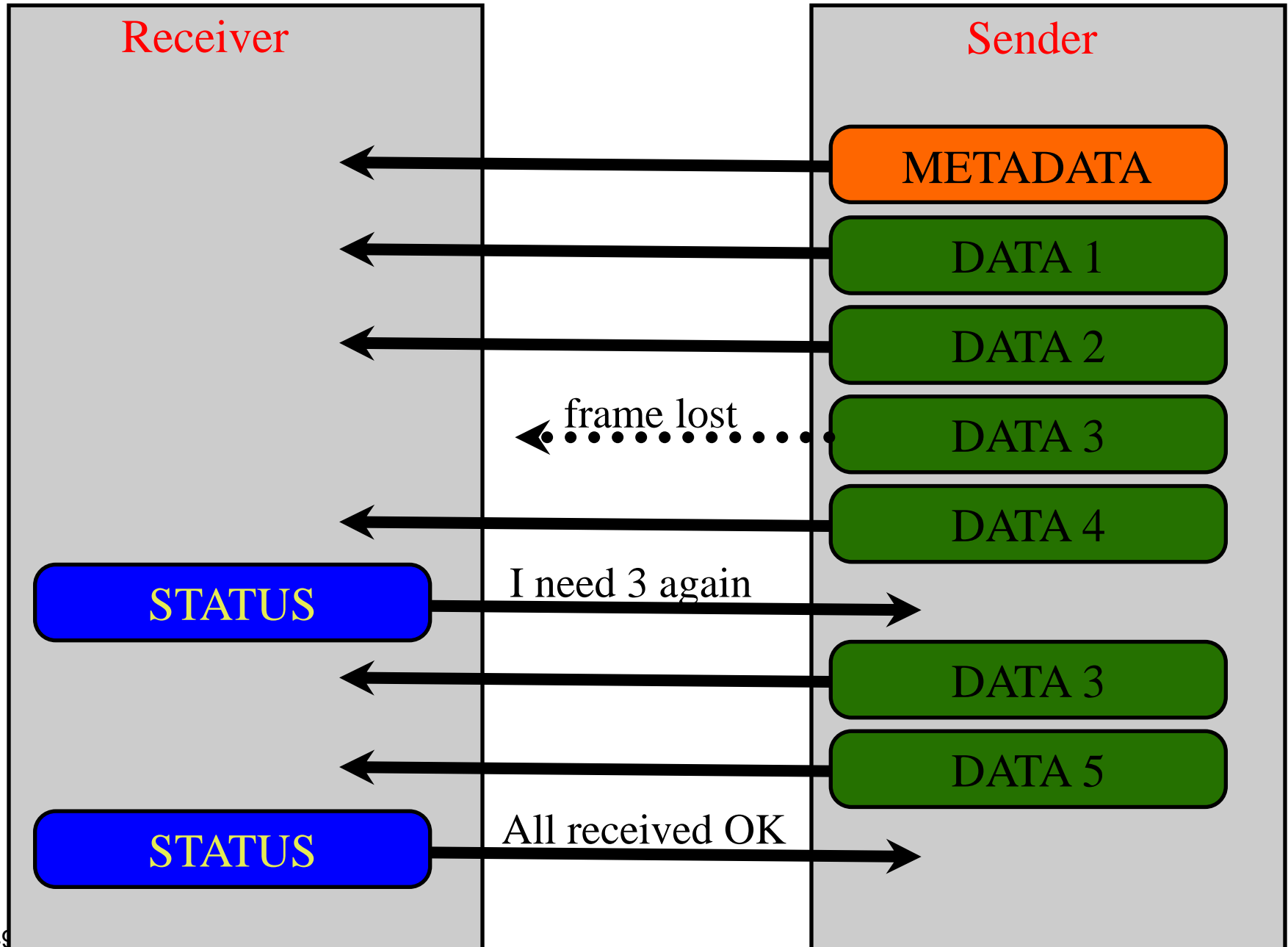
- **A suite of protocols specifically optimized for use on store-and-forward microsatellite communications missions.**
- **Design the spacecraft software so that there was one generic data entity that it knows how to deal with and deal with well**
- **Leave all routing decisions to the ground segment.**
- **A client identifies a message which it requires and sends a broadcast request to the server.**
- **The client will receive some or all of the datagrams from its request. Upon receiving these (or other) datagrams, the client places them at the indicated byte offset in the appropriate file.**
- **When a client wishes to have the server retransmit missing datagrams, the client composes and transmits a repeat request datagram. If some of the desired datagrams are not received, then the client sends the request again.**
- **The client repeats this process until the message is complete.**

# Generic SNACK GET Concept

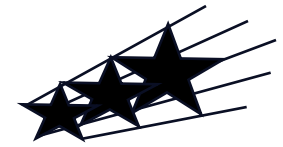




# Generic SNACK PUT Concept



# Selective, Negative Acknowledgment (SNACK) Protocols – not all inclusive



SYZYGY Engineering

## Consultative Committee for Space Data Systems (CCSDS) File Delivery Protocol (CFDP) - CCSDS 727.0-B-2:

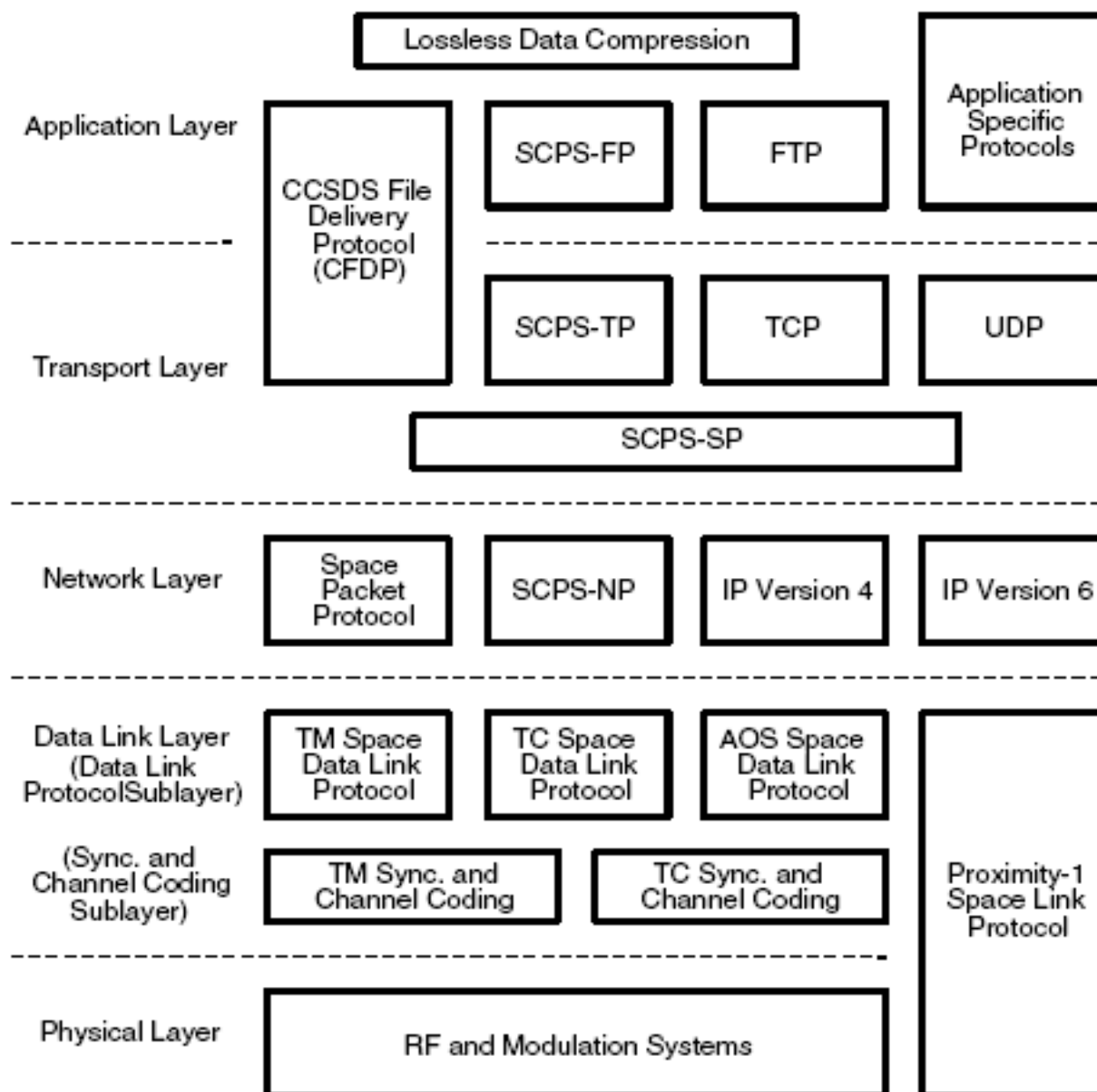
- Selective Negative Acknowledgement
- Designed for Space Applications (Highly Asymmetric Links, Suspend/Resume timers during known outages) **Highly stateful**
- Private Links (No Congestion Control)
- Designed for File Transfers (really an application and transport protocol)
- Reliable or Unreliable file transfer
- Routing Capabilities (Preconfigured Next Hop – Static Routing)

## Licklider Transport Protocol (LTP)

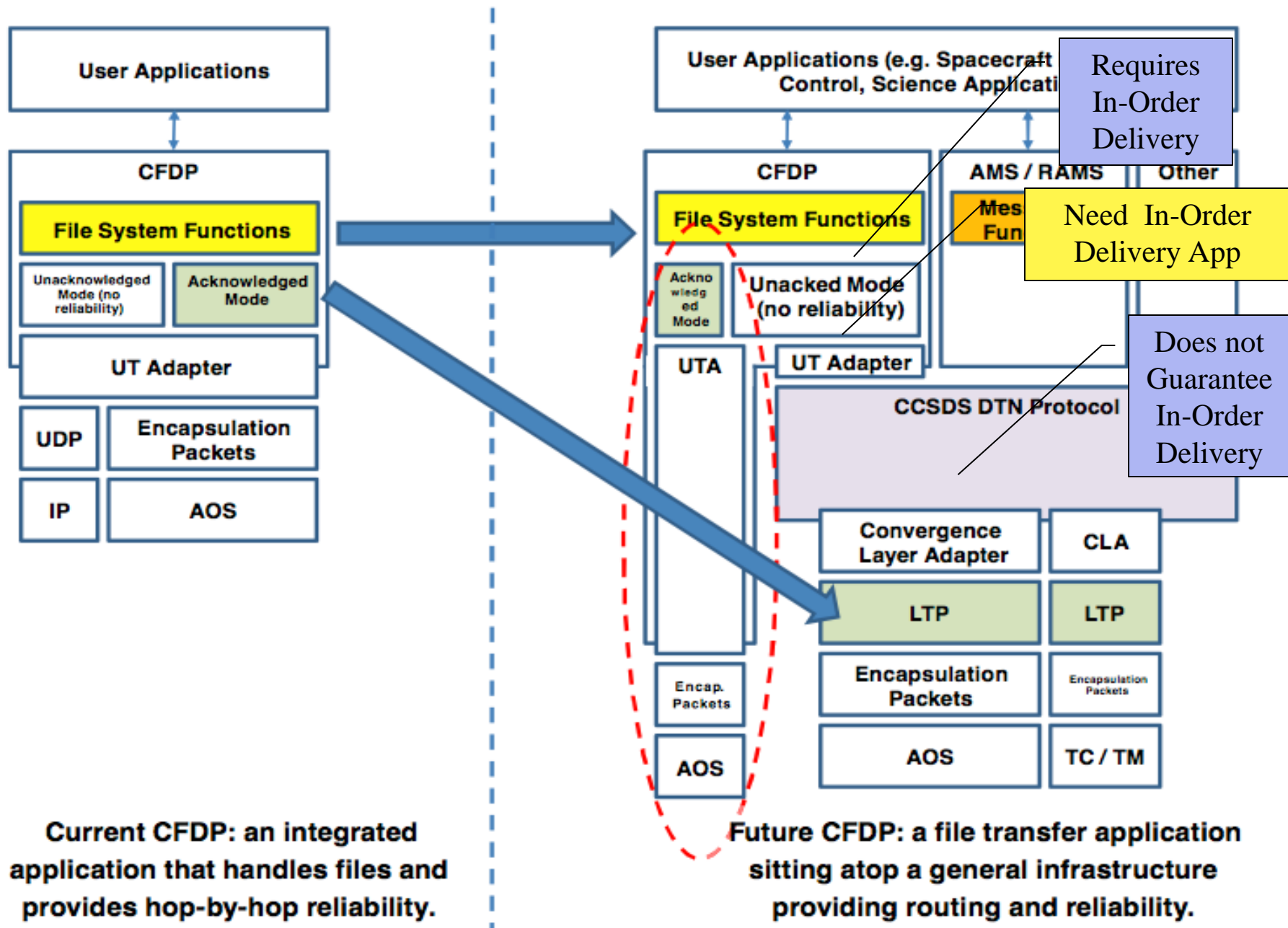
- Based on CFDP
- Distinguishes between important and unimportant data
- Timers work together with communication schedules and can be suspended whenever a scheduled link outage occurs
- Needs to be informed about link layer availability, round-trip time and communication schedule, basically requiring a management information base, Thus, **Highly stateful**

## Saratoga (used by SSTL daily)

- Based on “Protocols For Store-and-forward Message Switching Via Microsatellites”
- CFDP proved overly complex
- High-speed, UDP-based, peer-to-peer protocol, providing error-free guaranteed delivery of files, or streaming of data.
- No specified timers means no timeouts, so Saratoga is ideal for very long propagation delay networks (such as deep space)
- Able to cope with high forward/back network asymmetry (>850:1)

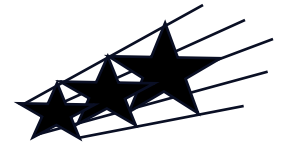


**Figure 2-1: Space Link Protocols**

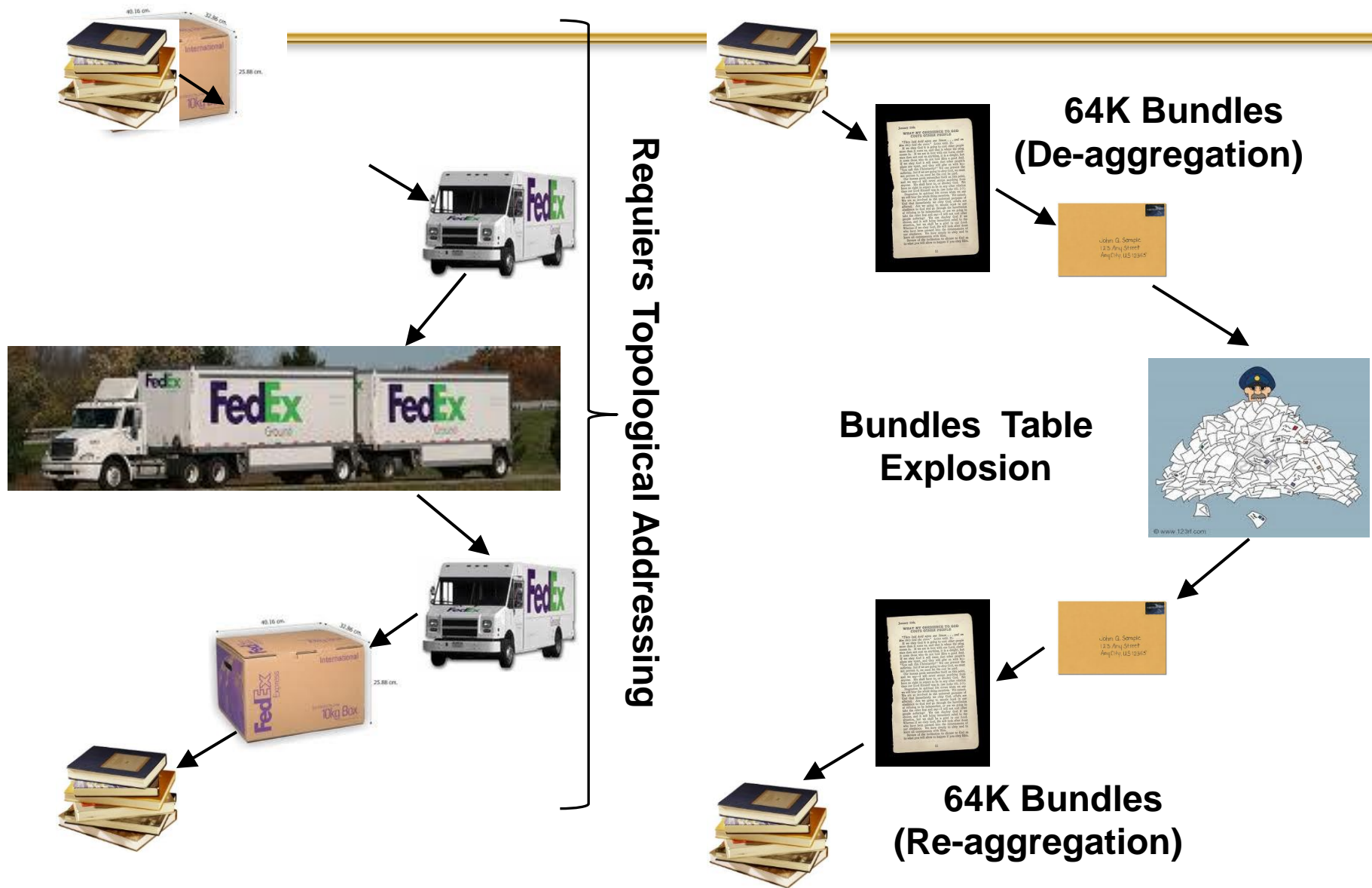


**Figure 4-4: A CFDP Evolution Path to Use DTN as the CFDP Unitdata Transfer Service**

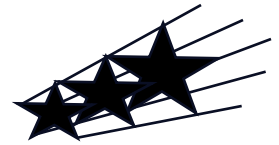
# Aggregation (or Not?)



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# Delay/Disruption Tolerant Network (DTN)



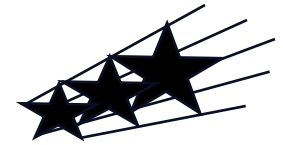
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- Goal: A standardized store and forward protocol and routing protocol
- Designed for extreme environments
  - Large transmission link delays
  - Extended periods of network partitioning
  - Routing capable of operating efficiently in the following environments
    - Frequently disconnected
    - Pre-scheduled or Opportunistic link availability
    - Heterogeneous underlying network technologies (including non-IP-based internets)
- The architecture operates as an overlay network

**Applications must be able to  
tolerate disconnection / disruption**



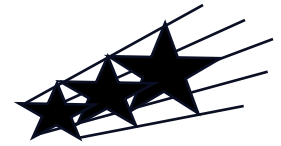
# Why DTN?



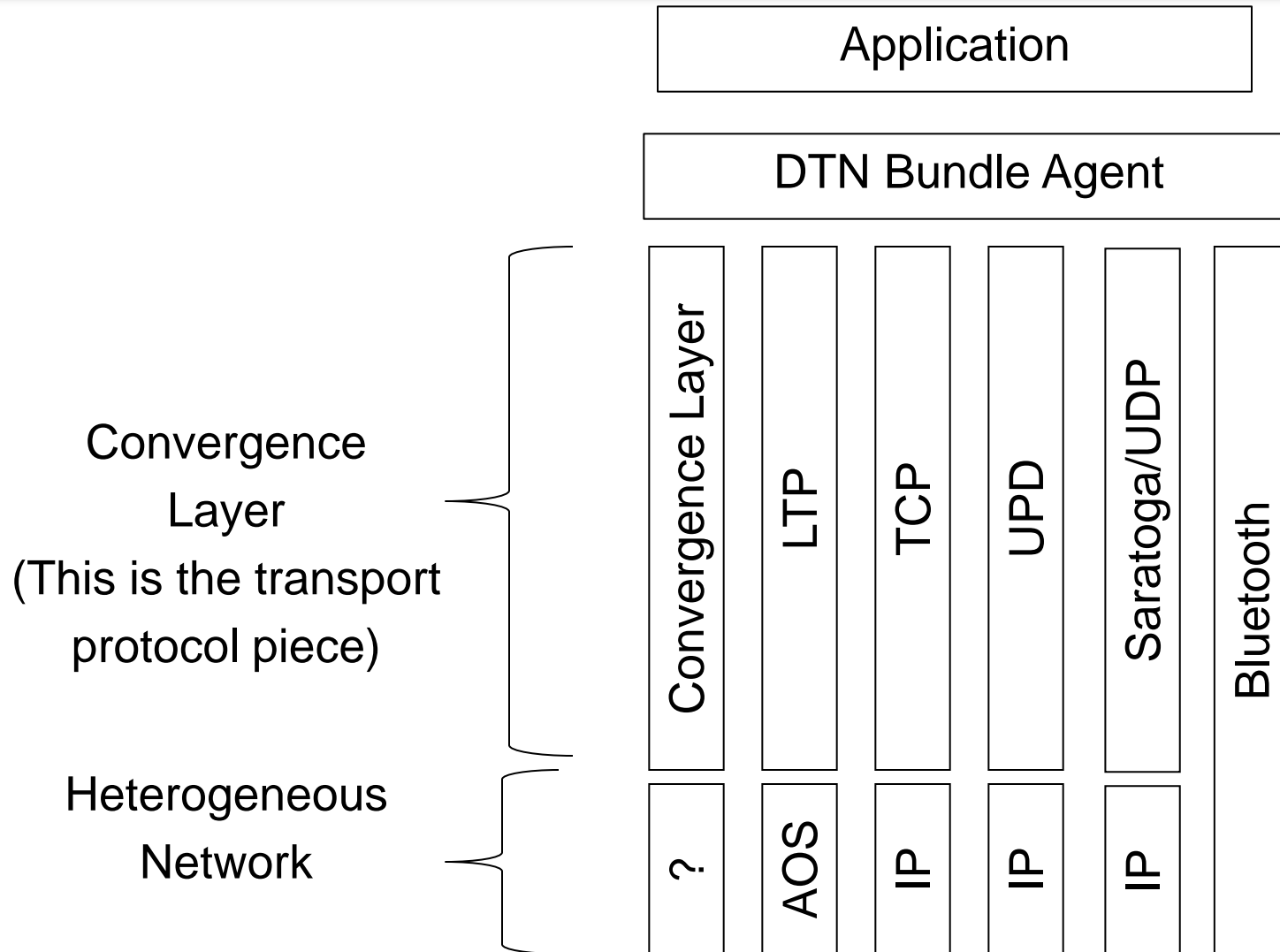
SYZYGY Engineering

- **Pros**
  - Flexibility
  - Reliability
  - Robustness (in that it can handle disconnection and disruption)
  - Automates routing for multi-hop, multi-path network topologies
- **Cons**
  - Added complexity
  - Not needed for single hop systems
  - Adds Overhead (~70 bytes)
  - Network overlay
    - Needs all the functions found in any network and more
      - Routing
      - Discovery
      - Name/Address Binding
      - Security
      - Network Management
      - Resource Management

# DTN Layering (RFC 5050)



SYZYGY Engineering

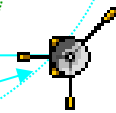
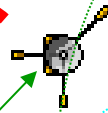
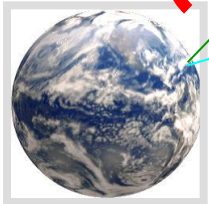


# DTN Environments

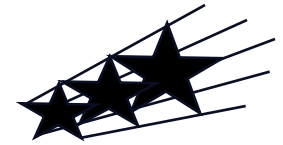
**Scheduled  
(long delay)**

**Opportunistic  
(intermittent  
connectivity,  
short delay)**

**Opportunistic  
(intermittent  
connectivity,  
short delay)**

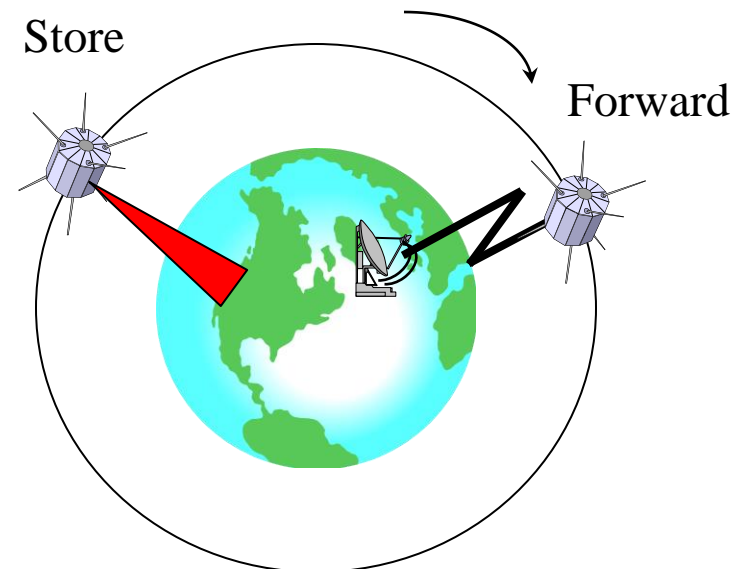


# Single-Hop Architecture

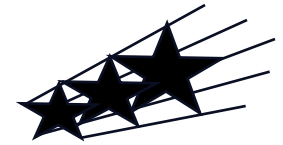


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- **CFDP**
- **LTP**
- **Saratoga**
- **Others (e.g., 9P)**
- **No Need for DTN Complexity**
  - DTN is most useful for multi-hop networks



# Known Issues with Current Bundling Specifications



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- Note: IRTF generated Request For Comments (RFCs) are standards.
  - RFC5050 is an Experimental Specification not a Standard
- Requires loose time synchronization
- No reliability checks on the header
- No way to terminate routing loops
- Naming is addressing
  - Flat name space makes scalability hard if not impossible
  - No identified way to aggregate routes
- Dynamic Routing is incomplete
- Discovery is incomplete
- Bundle Security Protocol
  - Specification is overly complex
  - Cases have been identified where it is broken
  - Security does not work with reactive fragmentation
  - Key exchange and key management is an open issue

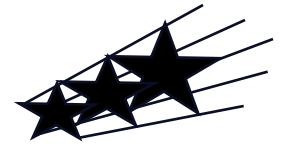


# DTN ON THE INTERNATIONAL SPACE STATION





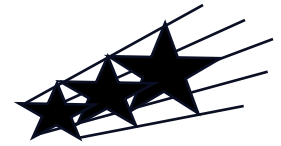
# ISS History



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- Design dates back to 80s and early 90s
  - pre-Internet era
- Phase 0
  - Commanding via low rate S-Band uplink
    - Uplink bandwidth is limited resource that must be managed.
  - Payload data downlink via Ku-Band at 50 Mbps
  - **Highly asymmetric links and both may not be on simultaneously**
- Phase I
  - KuBand uplink at 3 Mbps
  - Orbital communication adapter (introduced Internet
    - 3 Mbps Up, 6.5 Mbps down IP in CCSDS packets
- **KuBand Upgrade**
  - **30 Mbps Up, 300 Mbps Down Ku-Band**
  - **IP onboard with IP encapsulated in CCSDS up/down**

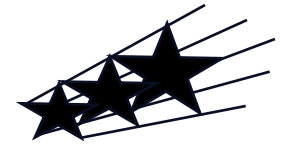
# NASA ISS DTN Project



SYZYGY Engineering

- ISS Disruption Tolerant Networking (DTN) Architecture for flight, ground, and test/simulation
- Increased reliability of payload data transfers between ISS and remote payload control centers during Acquisition of Signal (AOS) / Loss of Signal (LOS) transitions
- Increased automation of Payload Developer (PD) requests for data transfers
- Mechanism to alleviate extensive support to plan payload transfers around AOS/LOS and operator required transfers
- Mechanism to use standard, publicly available protocols, avoiding the use of costly custom protocol implementations
- Opportunity to gain valuable experience using DTN, which is the expected communication protocol of choice for future space exploration

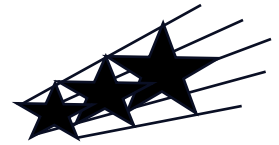
# Current Payload Operations



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- Uplink/Downlink request
- Entire files have to be re-transmitted when transfer errors occur
- Manual transfer by operations required for file transfers across AOS/LOS
- 24x7 continuous support operations to ensure access to science data
- Data recorded to Outage Mass Memory unit requires operators to playback
- Custom file transfer protocols utilized. Currently many payloads use a custom method or protocol to transfer data reliably between on-orbit and ground
  - Ex. Alpha Magnetic Spectrometer (AMS) and custom AMS Crew Operations Post (ACOP) system to store and buffer data in a proprietary implementation

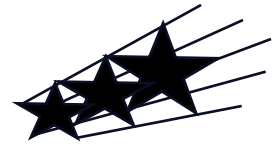
# DTN Benefits to Payload Communications



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- Provides capability to automate operations and ensure science delivery with little regard for link or facility outages
  - MSFC, MCC-H, and ISS DTN nodes will store user file uplinks/downlinks and forward bundles as Ku-Band becomes available
- Reduce real-time support to access and downlink science data
- DTN stores data during LOS and automatically initiates transfer upon AOS
- A download transfer can span Ku-Band AOS periods without any special scheduling or scripting
- Reduces need for duplicate storage and extra retrieval actions

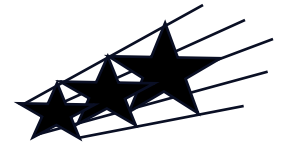
# DTN Benefits to Payload Communications



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- Reliable data transfer for ISS during LOS/AOS cycles
  - Automatic verification of bundle receipts, retransmissions reduced
  - When transmission errors occur only the bundles that have errors are retransmitted
  - Maximizes use of bandwidth by reducing the amount of data that has to be retransmitted
- Allows Payload Developers to use DTN protocols for their own applications
- Efficient use of downlink stream through DTN Quality of Service (QoS)
- Tolerance for high network latency (600 ms Round Trip Time (RTT) delay is typical on Ku link)

# Current/Future ISS DTN Use Cases



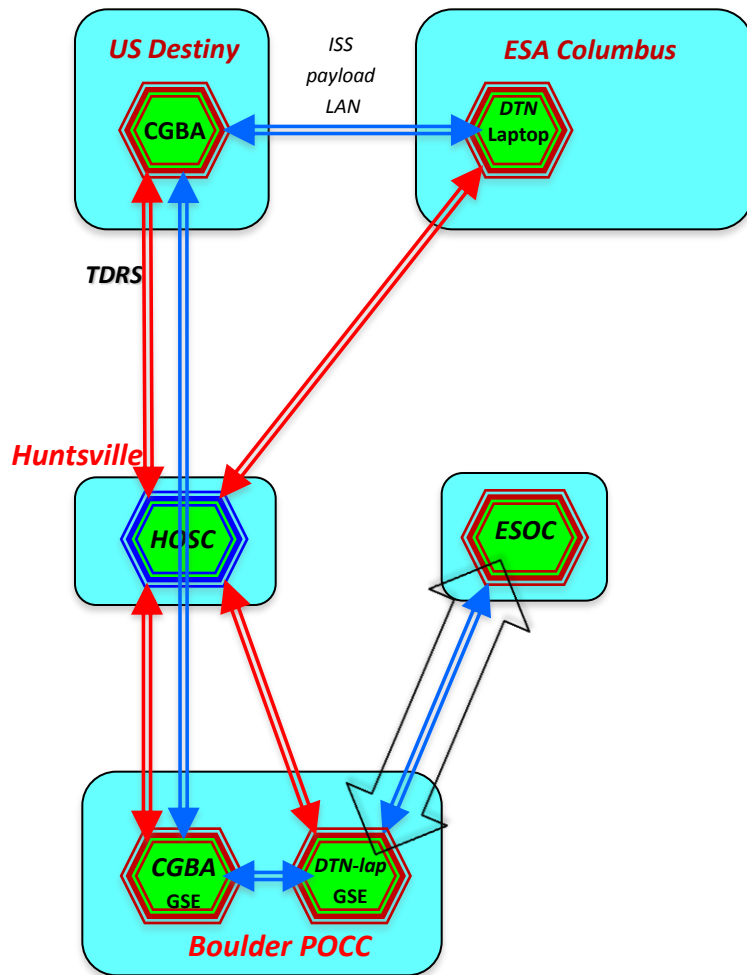
SYZYGY Engineering

- Current Users of DTN on ISS:
  - Various “developmental” DTN configurations have been in use on ISS in support of payload activities for several years
  - DTN is or has been used in support of the following Payloads
    - BioServe Commercial-Grade Bioprocessing Apparatus (CGBA)
    - DARPA Synchronized Position Hold Engage and Reorient Experimental Satellites (SPHERES) Smartphone
    - ESA Quickstart and OPSCOM I
    - ESA’s METERON (Multi-Purpose End-To-End Robotic Operation Network)
- Additional Payloads discussing use of DTN include:
  - Multiple User System for Earth Sensing (MUSES)
  - Human Exploration Telerobotics (HET) Surface Telerobotics
  - JAXA Kibo KaBand Rain fade mitigation
  - Other payloads are becoming interested

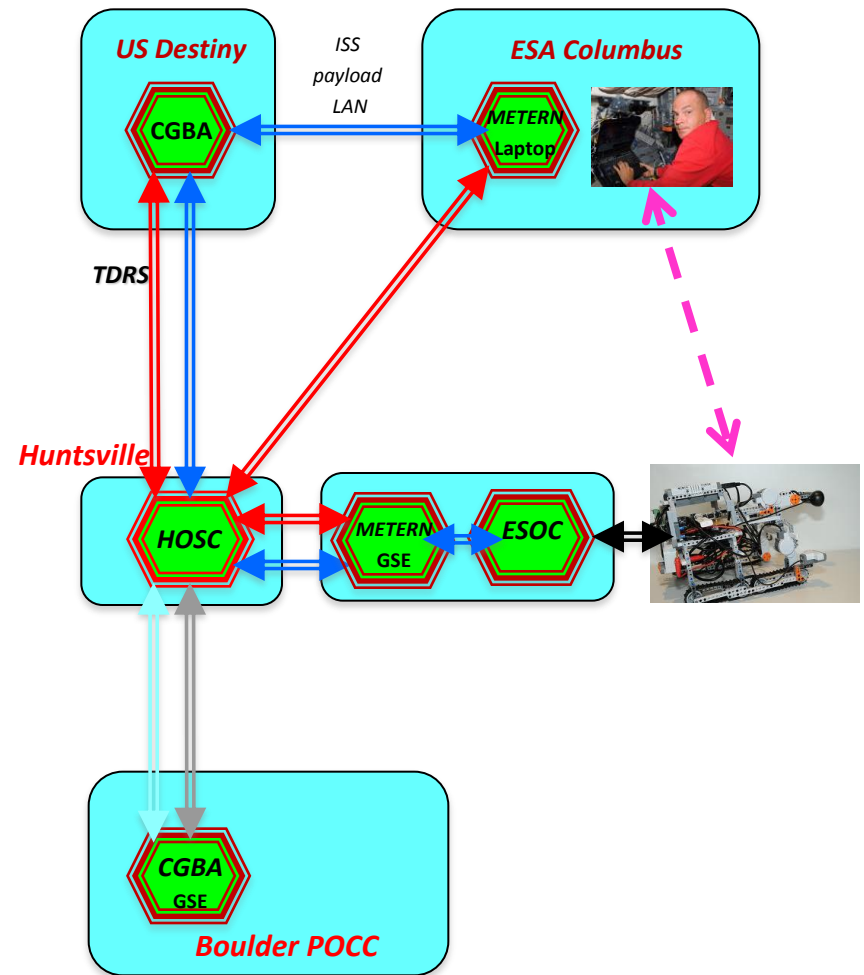


# ESA QuickStart-A and METERON topologies

## QuickStart-A



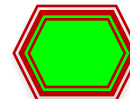
## METERON



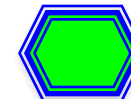
C&T /  
H&S



DTN

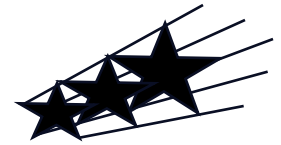


DTN  
Node

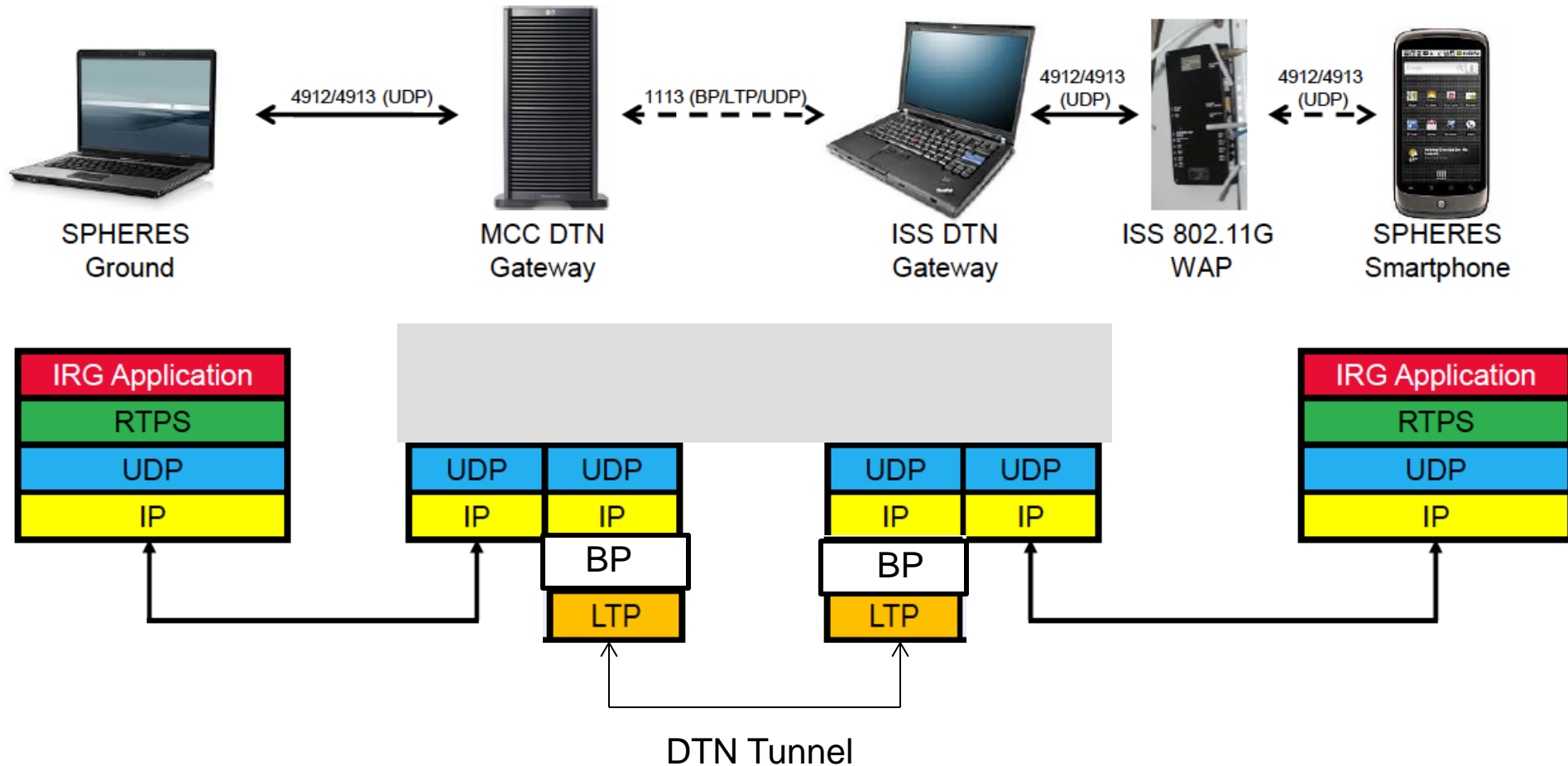


DTN  
Pass-  
through

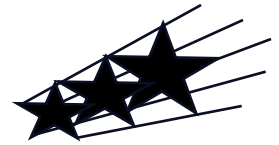
# Spheres



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# Summary



**SYZYGY Engineering**

- The implementation of DTN on ISS will provide a standard method of communication for payloads that is reliable, autonomous and more efficient than current techniques, resulting in better utilization and more science return from ISS
- The use of DTN by payloads will significantly ease Payload Development of both onboard and ground communication systems and could reduce payload operator costs